### STELLA MARY'S COLLEGE OF ENGINEERING

(Accredited by NAAC, Approved by AICTE - New Delhi, Affiliated to Anna University Chennai)

Aruthenganvilai, Azhikal Post, Kanyalumari District, Tamilnadu - 629202.

### **ME8593 DESIGN OF MACHINE ELEMENTS**

(Anna University: R2017)



#### Prepared By

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#### DEPARTMENT OF MECHANICAL ENGINEERING

#### **COURSE MATERIAL**

REGULATION	2017
YEAR	III
SEMESTER	05
COURSE NAME	DESIGN OF MACHINE ELEMENTS
COURSE CODE	ME8593
NAME OF THE COURSE INSTRUCTOR	Mr. J. STARLIN DEVA PRINCE

#### **SYLLABUS:**

#### UNIT I STEADY STRESSES AND VARIABLE STRESSES IN MACHINE MEMBERS 9

Introduction to the design process - factors influencing machine design, selection of materials based on mechanical properties - Preferred numbers, fits and tolerances – Direct, Bending and torsional stress equations – Impact and shock loading – calculation of principle stresses for various load combinations, eccentric loading – curved beams – crane hook and 'C' frame- Factor of safety - theories of failure – Design based on strength and stiffness – stress concentration – Design for variable loading.

#### UNIT II SHAFTS AND COUPLINGS

9

Design of solid and hollow shafts based on strength, rigidity and critical speed – Keys, keyways and splines - Rigid and flexible couplings.

#### UNIT III TEMPORARY AND PERMANENT JOINTS

9

Threaded fastners - Bolted joints including eccentric loading, Knuckle joints, Cotter joints –Welded joints, riveted joints for structures - theory of bonded joints.

#### UNIT IV ENERGY STORING ELEMENTS AND ENGINE COMPONENTS

9

Various types of springs, optimization of helical springs - rubber springs - Flywheels considering stresses in rims and arms for engines and punching machines- Connecting Rods and crank shafts.

UNIT V BEARINGS 9

Sliding contact and rolling contact bearings - Hydrodynamic journal bearings, Sommerfeld Number, Raimondi and Boyd graphs, -- Selection of Rolling Contact bearings.

#### **TEXT BOOKS:**

- 1. Bhandari V, "Design of Machine Elements", 4th Edition, Tata McGraw-Hill Book Co, 2016.
- 2. Joseph Shigley, Charles Mischke, Richard Budynas and Keith Nisbett "Mechanical Engineering Design", 9th Edition, Tata McGraw-Hill, 2011.

#### **REFERENCES:**

- 1. Alfred Hall, Halowenko, A and Laughlin, H., "Machine Design", Tata McGraw-Hill, BookCo.(Schaum's Outline), 2010
- 2. Ansel Ugural, "Mechanical Design An Integral Approach", 1st Edition, Tata McGraw-Hill Book Co, 2003.
- 3. P.C. Gope, "Machine Design Fundamental and Application", PHI learning private ltd, New Delhi, 2012.
- 4. R.B. Patel, "Design of Machine Elements", MacMillan Publishers India P Ltd., Tech-Max Educational resources, 2011.
- 5. Robert C. Juvinall and Kurt M. Marshek, "Fundamentals of Machine Design", 4th Edition, Wiley, 2005
- 6. Sundararajamoorthy T. V. Shanmugam .N, "Machine Design", Anuradha Publications,

#### **Course Outcome Articulation Matrix**

	Program Outcome											PSO			
Course Code / CO No	1	2	3	4	5	6	7	8	9	10	11	12	1	2	3
ME8097 / C426.1	3	2	0	3	0	0	0	0	2	3	0	3	3	3	0
ME8097 / C426.2	3	3	0	3	3	0	0	0	2	3	0	3	3	3	0
ME8097 / C426.3	3	3	0	3	3	0	0	0	2	3	0	3	3	3	0
ME8097 / C426.4	3	3	0	3	3	0	0	0	2	3	0	3	3	3	0
ME8097 / C426.5	3	3	0	3	3	0	0	0	2	3	0	3	3	3	0
Average	3	3	0	3	2	0	0	0	2	3	0	3	3	3	0

Type tion to the Design Process UNIT-2 60(5)

Design:

A plan or drawing Produced to show the LOOK and function or working of a building garment or other object before it is made.

(or)

Design is the Creation of Plan (or) conversion for the construction of an object or a system.

## Machine Design.

Machine design 95 defined as the use of scientific Principles, technical information and imagination in the description of a machine (or) a mechanical type System to perform specific functions with Maximum economy and Efficiency.

New ideas ? -> Design Process Designers Ideas

> Types of Design:

- 1. Adaptive Design
- 2. Developed Design.
- 3. New Design.

# 1. Adaptive Design:

1. Adaptive besign

Adaptation of existing designs with Minor Modification.

# 2. Developed cesign

Designer Starts from an existing Method.

### 3. New Design

Find outcome may differ

# => DESIGN Process

Recongnition of Need

Definition of Problem

Synthesis

Analysis and optimization

Evaluation

proentation.

\*Recongnition of Nead:

The First phase of activity which refers to the identification of demand for any design (or) Modification of any existing design.

\* Definition of Problem

It is nothing by but specifying the actual requirement and Present Stage of the Problem.

\* Synthesis: and Analysis

some group of activities is corried out on different models by changing like model formation, applying physical and technical Principles, computation and checking the result etc.

In this phase, a small model (Prototype) 15 formed and It is tested under various working conditions and the results are goothered for making final design.

### \* Optimitation:

This is the next step in which the above analysis is carried out on different models by changing the variable like moderial, loads, Methods, operatingtempole.

Evaluation 15 a significant phase of the total design Process. Evaluation 15 the final Proof of the Quies Successful design and usually involves the testing of a Prototype in the Laboratory.

### \* Presentation:

The Proper communication of design details like correct dimensions, Machining Process, tolerance details and other working condition.

# => Factors Influencing Machine Design:

- \* Type of Loading
- \* Size and Shape of the object.
- \* Material Properties required
- \* Environmental conditions
- \* place of usage
- \* Human Safety
- Cost
- \* Service Life
- \* Appearance
- \* Quantity Required
- \* Handling Provisions

mortifier facilities are

may methods.

\* Type of Loading

> Steady Load > dynamic Load > Impact Load

\* size and shape of the Object.

> small or big size, simple or intricate Shape

\* Moderial Properties.

buch as hard , soft , rigid , transpared opaque, conductive, ductile and brittle.

\* Environmental condition.

Such that the components 15 to be operated in corrective con non-corrisive atmosphere cool or hot conditions etc.

\* place of usage:

such that the machine is employed in land or waste on water or air (ie space) etc.

\* Human safety

For which the parts should have provisions for safe handling and easy maintenance etc.

\* cost

Cost is the another predominant factor for which machine component should be designed.

#### \* Service Factor Life:

For Long service Life the machine part should be very strong and for service Life, comparatively less strong item is sufficient.

\* Appearance

This Factor in so movely meant for fast sales Promotion.

\* Quantity Required This factor is decided by the Maderial and place for usage.

\* Handling Provisions: Any component shalld be designed buch that it must have some provisions for easy handling during shifting from one places and Proper Methods to another.

\* Workshop Facilities and Manufacturing Method: Unless there are suitable workshops in the nearb places and Proper Manufacturing method.

# => Selection of material based on Mechanical Properties

- \* strength
- \* Hardness
- \* Toughness
- \* Ductility
- \* Malleability
- \* Elasticity
- \* Plasticity
- \* brittleness
- \* stiffness
- \* Creep
- \* . Fatigue.
- \* Regilience

# \* strength

Ability of Material to applied load without failure.

### \* Hardness

Ability of a material to reach abracion.

# \* Toughness

Ability of Material to reach shock Loads.

at Dou

\* Ductility Property of Material which enable 15 to be

drawn into thin wires.

Property by which Moderial can be rolled in to thin sheets.

\* Elasticity

If any Maderial specimen 15 Loaded, its dimension changes, when the Load is removed, the specimen gels ititial shape.

\* plasticity

Plasticity 15 the property of a material which retains the deformation produced under Load permanently.

\* brittle ness

It is the Property of a moderial opposite to duefility.

The Property of breaking of a moderial

with little permanent distortion.

\* Stiffness

Load to the deflection.

\* Creep when a part 15 subjected to variable loads, at high temp for a long period of time, it will undergo a slow and permanent deformation is called Creep

\* Fatigue: (or) clastic limit

When any machine clement bubilected to Variable loads, it may fail before the stipulated time which has been calculated by treating the machine element under static or steady Load and that failure due to variable load is called fatigue failure.

### \* Resilience

In the ability of Maderial to resist absorb energy and to resist shock and impact Load.

### ⇒ Fits and Tolerances:

The

\* Fits

The degree of tightness or looseness blw two mating parts known as the fit of the Part.

This fits can be classified as running [(or)sliding] fit, drive fit, force fit, and shrink fit.

### \*Tolerance

The total Permitted variation in a Particle dimension of a single part in termed as the tolerance.

# \* Preffered Numbers

when a machine in to be made in Several size having dufferent power (or) Capaut If in necessory to decide what capacities will cover a certain range efficiently with min Number of 512 eg.

=> Direct Stress: In Machine Part

\* Load:

It is defined as any external force

acting upon a machine part

\* Types of Load

16

28

11

24

19

> Dead or steady Load

when it does not change in

magnitude as ord (or) direction.

+ Live (or) Variable Load

A Load is said to be a live or variable load, when it changes continually.

> Suddenly applied or shock Load

when it is suddenly applied

(or) removed.

> Impact Load

when it is applied some impact

initial velocity.

when some external system of forces or Loads act on a body, the internal force (equal a and opposite) are set up at various sections of the body, which resist the External force. This internal force per unit area at any section of the body is known as unit stress (or) simply a stress. (T)

Stress 
$$\sigma = \frac{P}{A}$$

P= Force (or) Load acting on a box A= cross-sectional area of the book

1

1 Pa = 1 N/m2

1 MPa = 1×10 N/m2 = 1N/mm2 1 GPa = 1×109 N/m2 = 1×103 N/mm2

When a force system of forces or Loads act on a body, it undergoes some deformation. This deformation per unit length is known as unit straver (or) simply a strain.

E = Change in Langth.

$$E = \frac{Sl}{l}$$

$$E_t = \frac{SL}{R}$$

$$\xi_{t} = \frac{81}{A}$$

Hooke's Law', States that when a material is Loaded within elastic Limit, the stress is directly

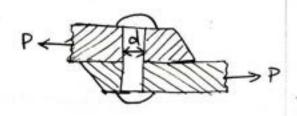
Propositional to Strain

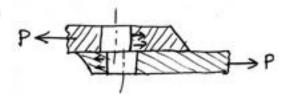
$$\sigma = E \times E$$

$$E = \frac{\sigma}{E} = \frac{P/A}{8UIL}$$

$$E = \frac{pl}{A \times 8l}$$

When a body is subjected to two equal and opposite force acting tangentially accross the resisting section, as a result of which the body tends to shear off the section, then the stress induced 15 called shear stress.





The corresponding Strain is known as shear Strain and It is measured by the angular determat accompanying the shear stress.

A= Tyux d2 [Area of hole]

$$T = \frac{P}{A} = \frac{TP}{T_{/4} \times d^2} = \frac{4P}{T d^2}$$

Double Shear blress

$$Z = \frac{P}{a \times A} = \frac{P}{2x \frac{T}{4} \times d^2} = \frac{2P}{T d^2}$$

A= Tide [Area of Plate]

# > Shear Modulus (or) Modulus of Rigidity

It has been found experimentally that with in the to elastic limit, the shear stress is directly Proportional to shear strain

Zap

7 = c p

. \_ C T

\$ = 6 hear strain.

c= 1

# > working stress

when designing machine parts, it is describle to keep the stress Lower than the maximum or ultimate stress at which failure of the material take place.

This stress known as working stress or Design stress.

# -> Factor of safety.

As the ratio of the maximum stress to the working stress.

FOS = Maximum stress.

Working stress.

When a machine member is subjected to the action of two equal and opposite couples acting in parallel planes (or torque or twisting moment), then the machine member is said to be subjected to torsion. The stress set up by torsion 15 known as torsional shear stres

J= 1/32 XD4

T= Torsional Shear Strees

Tmax= T/16 x Txd3

r = Radius of Shatt

P= 2 TH NT

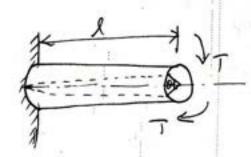
J= polar moment of inertia

C= Modulus of Rigidity

l: Length of the shaft

T= Torque or twisting Moment

0 = Angle of twist in radius on a length &



### >Bending Stress

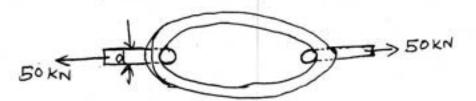
The machine part of structural members may be subjected to static or dynamic Load which cause bending stress in the sections besides other types of stresses such as tensile, compressive and shearing stresses.

$$\frac{M}{T} = \frac{\sigma_b}{y} = \frac{E}{R}$$

# Problem based on Direct 6tress, bonding stress (10)

Torsional stress:

1. A coil chain of a crane required to carr a maximum Load of 50KN is shown in Fig.



Frnd the diameter of the link stock, if the permiss textile stress in the link material is not exceed 75MPE

### Goven:

Load P= 50 KN = 50x103 N

bensile stress of = 75 MP9 = 75 XIOB N/m2.

5t = 75 N/mm2

To Find

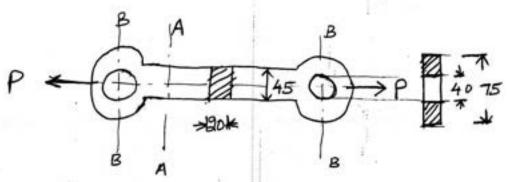
diameter of the Link d

Soln

Stress 
$$\sigma = \frac{\rho}{A}$$

2. A cast Iron Link, as shown in Fig 4-4 15 n
required to transmit a steady tensile Load of 45km.

Find the tensile stress induced in the link
material at sections A-A and B-B.



315, 10,11,

15, 27, Given:

3084,305, Tensile Load P=45KN =45X103N

Torind Tensile Stress at A-A

Tensile Stress at B-B

 $\frac{50ln}{\text{Tensile}}$  Hrees at section A-A  $\sigma_t = \frac{p}{A} = \frac{1}{4}$ 

A1= Area at section A-A

A1= 20×45 [Reitangle setton bd]

A1= 900 mm2

Tensile stress induced at section B-B

$$O_{\frac{1}{4}} = \frac{P}{A_2}$$

Az : Area of section &B

3) Ahydraulic press exerts a total Load of 3.5 MN. This Load is carried by two steel rods, supporting a upper head of the press. If the safe stress is 85 Mpa and E= 210 KN/mm² Find I. diameter of the rods, and 2. extension in each rod and a Length of 2.5 m.

Given:

Load 
$$P=3.5\text{MN}=3.5\text{X} \cdot 5\text{N}$$

Stress  $T=85\text{MPa}=85\text{X} \cdot 6\text{N}/m^2$ 
 $T=85\text{N}/m^2$ 

Young's Modulus  $E=210\text{X} \cdot 10^3\text{N}/m^2$ 

Length of rod  $L=2.5\text{m}$ .

### To Find:

- 1. Drameter of the rods d
- 2. Extension in each rod Sl

### 50 n

$$P_1 = \frac{P}{2} = \frac{3.5 \times 10^6}{2}$$
 $A = \frac{11}{4} \times d^2$ 

$$\frac{11/4 \times d^2}{85 = \frac{1.75 \times 10^6}{11/4 \times d^2}}$$

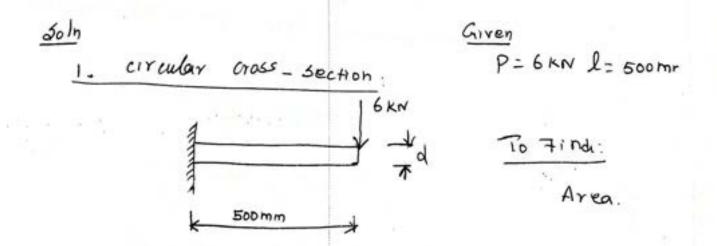
# 2. Extension in each rod Sl

$$\delta l = \frac{\sigma \times l}{E} = \frac{85 \times 2.5 \times 10^3}{210 \times 10^3}$$

4. A cartilever of Span 500mm Carries a vertical down (1)
Ward load of 6 kN at its Free end. Assume
Yield value of 350 Mpa and Factor of safety
as 3. Find the Economical Section for Cantilever
among

a. circular cross-section of drameter d' b. Rectangular cross section of depth h' and Width t with hlt=2

c. I section section of depth 7t and flange width 5t where t is the thickn specify the dimension and cross sections area of the economical section.



Load = P = 6kn = 6x103N

Length l = 500mm

Yield Stress 0 = 350MPa = 350N/mm2

Fas n = 3

For bed bending Eqn
$$\frac{M}{I} = \frac{S}{y} = \frac{E}{R}$$

$$\frac{M}{I} = \frac{0}{y}$$

$$y = \frac{d}{2}$$

$$\frac{3 \times 10^{6}}{\sqrt{10}/64 \times d^{4}} = \frac{116.7}{d/2}$$

$$\frac{61.115\times10^{6}}{d^{3}} = \frac{116.7\times2}{}$$

$$\frac{\sqrt{2}}{2} = \frac{\sqrt{2}}{3}$$

b. For Rectangular section of depth h and width t

$$\frac{M}{I} = \frac{D}{\lambda}$$

$$T = \frac{4 \times h^3}{12} = \frac{4 \times (2 + )^3}{12} = \frac{4 \times 8 + 4}{12}$$

$$y = \frac{h}{2} = \frac{xt}{x}$$

$$\frac{3\times10^{6}}{8t^{4}} = \frac{116.7\times}{t}$$

$$\frac{3 \times 10^{6} \times 12}{8 \cdot t / 3} = \frac{116.7}{t}$$

$$\frac{1}{4} \cdot 5 \times 10^{6} = 116.7$$

$$\frac{1}{4} = \frac{14.5 \times 10^{6}}{116.7}$$

$$\frac{1}{4} = \frac{38.560.41}{116.7}$$

$$\frac{1}{4} = \frac{38.560.41}{116.7}$$

$$\frac{1}{4} = \frac{38.560.41}{116.7}$$

$$\frac{1}{4} = \frac{1}{4} = \frac{116.7}{116.7}$$

$$\frac{1}{4} = \frac{116.7}{116.7}$$

$$\frac{1}{$$

I = 1215t4

$$\frac{3\times10^{6}}{\frac{1215t^{4}}{12}} = \frac{116.7}{3.5t}$$

$$t^3 = 888.61$$
 $1 = 9.61 \text{ mm}$ 

Area = 
$$[5t \times 7t] - [4t \times 5t]$$
  
=  $35t^2 - 20t^2$   
Area =  $15t^2 = 15 \times 9.61^2$ 

**B**:

# ⇒ Impact & shock Loading

Machine members are subjected to the Load with impact. The stress Produced in the member due to the falling Load is known as impact stress.

over a period ob

Johns reached Suddenly 1. Gradually Load which with Conerd with Conerd and Conerd with Conerd and C C 2. Suddenly applied (or) impact or shock

which with Consider a bar carrying a Load Wat a height h. and falling on the collar provided at the

Lower end as shown in fig.

A= Cross-sectional Area of the bag

E = Young's Modulus

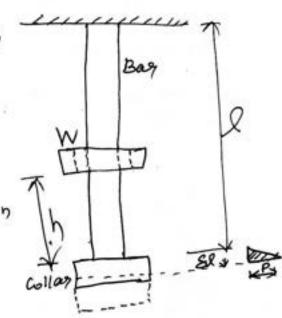
l : Length of the bar.

Sl: Deformation of the bar

P = Force at which the deflection SI is produced,

Ji = stress induced in the bag due to impact Load!

h = Height through the load talks



Kinetic at energy goined by the system of Strain energy = 1/2 xPx Sl Due to the Sallry bal the Energy is gained, weight of the same Potantial energy lost by the time the weight w' = W [h+ Se] falls down ( : Potential energy is (ost) Kinetic Energy = Potential Energy 1/2 × P × 8l = W[h+ 8l] J= 1 = P= 5 1 × 51 × A × 20 5 × = W[h+82] 1 × Si × A × ST; X = W [h+ Sixl] 5- 8×E => & XE Al (01) = wh + wl xo;  $\frac{Al}{\partial E} (\sigma_i)^2 - Wh - \frac{Wl}{E} \times \sigma_i = 0$ From this quadratic Eqn, we find and that ie, o; = o + o [1+ an] J=W/A €= e

Relation blw suddenly applied and gradually applied Load

Putting 
$$h=0$$

$$O'_{i} = \frac{2w}{A}$$

When the same Load W is gradually applied the stress produced

For this, it is clear that the stress due to suddenly load is double that of the gradually applied load

### Problem

An Unknown weight falls through 10mm on a collar rigidly attached to the Lower end of the Vertical bar 3m long and 600mm² in section.

If the maximum instantaneous extension is known to be 2mm, what is the corresponding stress and the value of unknown weight? Take E=200kN/mm²

height h= 10mm

Length 1 = 3m = 2000 mm

Area  $A = 600 \,\mathrm{mm}^2$ 

El = 2mm

Young's Modulus E= 200 KN/mm2 = 200X103 N/mm2

### To Find:

1. 5tress o

2.weight w

# <u>áoln</u>

1. Stress o

$$\nabla = E = \frac{\text{5+ress}}{\text{5+rain}} = \frac{\sigma}{E}$$

$$\nabla = E \times E$$

$$\nabla = E \times \frac{SL}{L} = \frac{200 \times 10^3 \times 2}{3000}$$

$$\nabla = 133.3 \, \text{N/mm}^2$$

# 2. Weight: (w)

W.K.T 
$$\sigma = \frac{W}{A} \left[ 1 + \sqrt{1 + \frac{2hAE}{Wl}} \right]$$

Potential Energy = kinetic Energy

W[h+8] = 1/2 x Px Sl

P = static Load = stress x Area

P = 133.3 x 6 0 0

P = 79.98 x103 N

W[10+2] = 1/2 × 79 × 98×103×2

W = 79 980

W = 6665 N

2) An I section beam of depth 250mm is supported at two points 4m apart. It is loaded by a weight of 4kN falling through a height h and striking the beam at mid span. Moment of thertia of a section is  $8 \times 10^7 \, \text{mm}^4$ . Modulus of elasticity is  $210 \, \text{KN/mm}^2$ . Determine the permissible value of h if the stress is limited to  $|20 \, \text{N/mm}^2$ .

Le Goven Data.

Length of beam l=4m=4000mmDepth of beam d=250mmImpact Load (falling)  $w=4kN=4x10^3N$ 

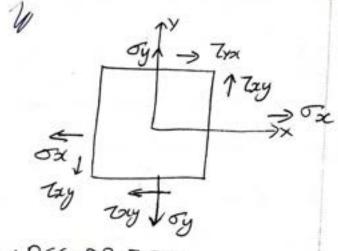
### To Find

when a system of forces act on a body, all the particles of that body are disturbed and their dimensions and Locations are varied due to straining action by the force.

Planes mutually Perpendicular to each other which carry only normal stresses and no shear stress.

These planes which are having only stress are called Principal planes and these normal stress are called Principal planes and these normal stress are called Principal planes stresses.

Among these principal stress some is having maximum value and other is having minimum value. There are some planes which are 45° to Principal planes and carry only maximum shear stresses and hence known as shearplane.

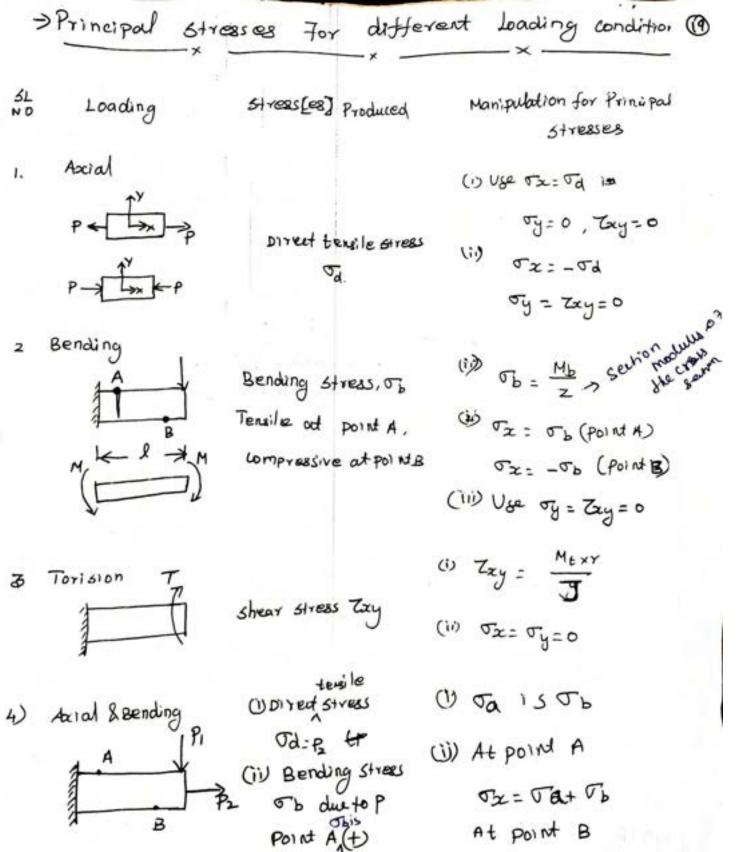


LPSG DB 7.27

Maximum Principal Stress 
$$\sigma_i = \frac{\sigma_{x+\sigma_y}}{2} + \left[\frac{\sigma_{x+\sigma_y}^2}{2}\right] + \left[\frac{\sigma_{x+\sigma_y}^2}{2}\right]$$

Minimum Principal Stress 
$$\sigma_2 = \frac{\sigma_2 + \sigma_y}{2} - \frac{\sigma_z + \sigma_y}{2} + \frac{\sigma_z + \sigma_y}{2} + \frac{\sigma_z + \sigma_z}{2}$$

(ex) 
$$\sigma_{a} = \frac{1}{2} \left[ \sigma_{x} + \sigma_{y} \right] - \sqrt{\left( \sigma_{x} - \sigma_{y} \right)^{2} + \frac{4}{7} \sigma_{y}^{2}}$$



Point 8 (

02 = Va -06

5 Aboral & Torsion

- (1) Direct tensile stress (i) of by Tayfind

  To due to P (ii) Ta=Td, Ty=0

  To Mt
- 6. Bending Ktorsion
- (1) Bending stress of (i) find of and Txy

  due to P (At AH (i) At point A ox: of

  1) tension, B at point B ox: or b

  it is compressive (iii) oy = 0
- 7. Axial bending
- (i) To due to P2 (Tensile)
- (1) Atpoint A

  Tz= Ta+Tb (both
  tensile)

- Torsion P1
- (i) To due to P

  [Tensile at A,

  compressive at B]

  (11) Try due to T
- (i)) oy=0 A to 1 At B 1) oz= oa - ob

(i) Tay is Present

[Ob 1500 Present (1) Ty = 0

NOTE:

1 Find Direct/Bending/Shear Stress as per the localing given.

- (5) Algebraichy and Direct Stress (Jay) & Bending stress (Ja)

  Ox 57+56

  Ox 57+56
- 3 Treat boy separatly. Do NOT Added with od (en) To.
- (2) use on and try in the principal somew Pan

Mb = 80N-1

1. A hollow shaft of 40mm owder diameter and 25 mm inner diameter is subjected to a twisting moment of 120 N.m., simultaneously, it is subjected to an axial thrust of 10 KN and a bending moment of 80 Nim Calculate the maximum compressive and shear stress

## Given

Outer dia at Shaft do = 40mm

Inner dia of shaft di = 25mm

Twisting Moment T = 120 N. m = 120x10<sup>3</sup>

Axial Load P = 10 KN = 10x10<sup>3</sup>N

Bending Moment Mb = 80 N·m = 80x103 N·mm

To Find:

- 1. Maximum compressive stress (52)
- 2. Shear Stress Tay.

Soln

1. Maximum compressive stress of

$$\sqrt{3} = \frac{3x + 3y}{2} + \sqrt{\left[\frac{3x + 3y}{2}\right]^2 + 7xy^2}$$

LDB.7.2>

OY

From table 
$$\sigma_{z}$$
:  $\sigma_{d}+\sigma_{b}$  Tensile.

 $\sigma_{z}$ :  $\sigma_{d}-\sigma_{b}$  compressive

Jd: Direct compressive stress. due to accial Logi.

$$\begin{aligned}
\nabla x &= \nabla d + \nabla B
\end{aligned}$$

$$\nabla d &= \frac{P}{A} = \frac{P}{\sqrt{4} \left[ do^2 - di^2 \right]}$$

$$(7.17)$$

To = bending stress due to bending.

Moment of meta I = 7/64 ×[do4-di4]

$$y = \frac{do}{2} = \frac{40}{2}$$

To Find Zocy

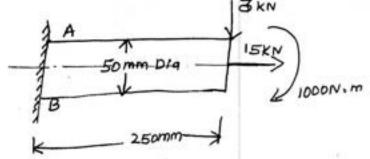
$$\frac{M_t}{I} = \frac{Co}{l} = \frac{Z}{r}$$

$$\gamma = \frac{do}{2} = \frac{40}{2} = 20mm$$

Result.

Emax = 10.51 N/mm2

A shaft as shown in 719. is subjected to a 2 bending Load of BKN, pure torque of 1000Non and an axial Pulling Force of 15 KN. collulate the stress at A and B



Given

Bending Load \* Distance = Pb = BKN = BXIDAN

Porque (or) Twisting Madered My = 1000 N.m =108x103 Nimm

Axial Load P = 15KN = 15 XIB3 N

Drameter of shaft d = 50mm

length of shaft 1 = 250mm

To Find:

The Principal Stresses at A & B

doln

At point A

7.17 
$$\sqrt{d} = \frac{P}{A} = \frac{15 \times 16^{2}}{\sqrt{1}/4 \times d^{2}} = \frac{16 \times 16^{2}}{\sqrt{1}/4 \times (50)^{2}}$$

$$\angle 7.1$$
  $\frac{Mb}{I} = \frac{\sigma_b}{y}$ 

$$y = d/2 = \frac{50}{2} = 25 \text{ mm}$$

Zmax

Maximum Shear stress Track

$$7max = \frac{1}{2} \sqrt{(5x-5y)^2 + 47yy^2}$$

$$= \frac{1}{2} \times \sqrt{(68.75-0)^2 + 4x(40.74)^2}$$

$$= \frac{1}{2} \times \sqrt{(68.75)^2 + 4(40.74)^2}$$

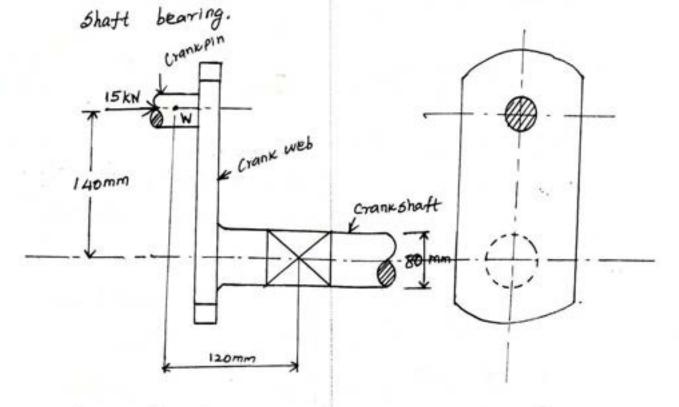
$$= \frac{1}{2} \times 106.60$$

$$7max = 53.30 \text{ N/mm}^2$$

$$\sigma_2 = \frac{1}{2} \left[ (-53.47) - \sqrt{\left[ -53.47 - 0 \right]^2 + 4 (40.74)^2} \right]$$

A tangential Load of 15 km acts on the Crank Pin.

Determine the maximum principal stress and the
maximum shear stress at the centre of the crank



Given data:

Tangentrafformal Load P= 15KN = 15X103 N

drameter d= 80mm

Vertical height h= 140mm Length (horizondal) 1= 120mm

## To Find:

- 1. Maximum Principal Stress O,
- 2. Maximum Shear Stress Tmax.

Mb= # Bending Moment at the centre of crankshaft bearing

$$7xy = \frac{Mt \times r}{3}$$

$$= \frac{2.1 \times 10^{6} \times 40}{4.02 \times 10^{6}}$$

$$M_t$$
 = torque transmitted to the Axial of the shaft  $M_t$  =  $P_X h = 15 \times 10^3 \times 140$   $M_t$  =  $2.1 \times 10^6 N.mm$ 

Any [ Zmax = 27.51 N/mm2

4. The stress due to state in a machine member is given as follows.  $\sigma_{z}=20\text{Mpa}$ ,  $\sigma_{y}=7\text{Mpa}$ ,  $\sigma_{z}=4\text{Mpa}$ , Find the Principal normal and the stresses. Locate the angle of  $\sigma_{z}$  and  $\sigma_{z}$  from x axis.

H.W

## => Theories of Failure: 27.3>

A given mathine member may fail [ie, it will no longer be able to Perform its Intended Function] due to various reasons in various methods modes.

It necessary to known the various conditions of failure of manhine members. some failure theores as Follows.

(1) Maximum Principal Stress [Ranxine theory]

Normal on the member of or \$\int\_2\$ or \$\int\_3\$ [Which is Maximum] = \$\inty\$

For design Purpose

(7) or \$\int\_2\$ or \$\int\_3\$ [Which is Maximum] = \$\inty\$ quitle Maximum] = \$\inty\$ ductile Maximum] = \$\inty\$ ductile Maximum] = \$\inty\$ ductile Maximum] = \$\inty\$ or \$\inty\$ or \$\int\_3\$ [Which is Maximum] = \$\inty\$ ductile Maximum] = \$\inty\$ ductile Maximum] = \$\inty\$ ductile Maximum] = \$\inty\$ do not \$\inty\$ for \$\inty\$ principal ductile Maximum] = \$\inty\$ or \$\inty\$ and \$\inty\$ do not \$\inty\$ ductile Maximum] = \$\inty\$ for \$\inty\$ ductile Maximum] = \$\inty\$ ducti

(ii) Maximum shear theory [Guest's or Couloum b's]  $(x) (\sigma_1 - \sigma_2) (\sigma_1) (\sigma_2 - \sigma_3) (\sigma_1) (\sigma_3 - \sigma_1) = \frac{\sigma_2}{n}$ 

(iii) Maximum strain Theory (or) [st Vanand's theory] (st)

oi - 2 (52+53) (or) oz - 2 [53+5] (or) oz -2 [5 1+52) = 7

person's rather

(iv) Maximum Strain Energy Theory

(v) Octahedral (ex) Distortion Energy Theory [von unses-Hencky]  $\sigma_1^2 + \sigma_2^2 + \sigma_3^2 - \sigma_1\sigma_2 - \sigma_2\sigma_3 - \sigma_3\sigma_1 = \sigma_3^2$   $\mathcal{D} = Poisson's Ratio$ 

## Problem 1

- 1. The Load on a bolt consists of an accial Pull of IOKN together with a transverse shear force of 5kN. Find the diameter of bolt required according to.
  - 1. Maximum Principal Stress Theory
  - 2. Maximum shear stress Theory
  - 2. Maximum Principal strain theory
  - 4. Maximum Strain chergy Theory
  - 5. Maximum distortion energy Theory

Take Permissible tensile stress at clostic Limit = 100 MPa and poissons ratio =0.3

# Given data: 1 OKN Axial tensile load Pt = lokn= lox103 N Transverse Shear force Ps = 5kn=5x103 N VOKN Permissible tensile stress $\sigma_t = 100 \text{MPa} = 100 \times 10^6 \text{ N/m}^2 = \frac{100 \times 10^6}{(10^3)^2} \text{ N/m}^2$ of = 100 N/mm2 Poisson's Ratio 2) or /m = 0.3 To Find: Diameter of Shaft Soln 1. Maximum principal stress Theory 07, =/1 [ (02+04) + (02-04)2 + 4 7xy2 <7.1> An Axral tensile stress ox = 0; $\sqrt{2} = \frac{P_4}{A} = \frac{10 \times 10^3}{71/4 \times 4^2}$ 12.73 x10 02= 07 03 = 03 图:

1. Maximum Principal Strees Theory 27.3>

$$\sigma_{1} = \frac{1}{2} \left[ (\sigma_{x} + \sigma_{y}) + \sqrt{(\sigma_{x} - \sigma_{y})^{2} + 4 \sigma_{y}^{2}} \right]$$

Ox = Stress due to tensile Load

$$\sqrt{2} = \frac{P_4}{A} = \frac{10 \times 10^3}{7 \times 10^2}$$

$$\sigma_{x_{2}} = \frac{12.73 \times 10^{3}}{d^{2}}$$

Z: Stress due to shear Load

$$Z = \frac{P_3}{A} = \frac{5 \times 10^3}{7 M_0 \times d^2}$$

$$Z = Z_{xy} = \frac{6.365 \times 10^3}{d^2}$$

$$\sqrt{1 - \frac{1}{2}} \left[ \left( \frac{12.73 \times 10^3}{d^2} + 0 \right) + \sqrt{\left[ \frac{12.73 \times 10^3}{d^2} - 0 \right]^2 + 4 \times \left( \frac{6.365 \times 10^3}{d^2} \right)^2} \right]$$

$$\sigma_{1} = \frac{12.73 \times 10^{3}}{2 d^{2}} + \frac{1}{2} \int \left[ \frac{12.73 \times 10^{3}}{d^{2}} \right]^{2} + 4 \left[ \frac{6.366 \times 10^{3}}{d^{2}} \right]^{2}$$

$$\sigma_1 = \frac{6.365 \times 16^3}{d^2} + \frac{1}{2} \times \frac{6.365 \times 10^3}{d^2} \sqrt{2^2 + 4}$$

$$\begin{aligned}
\nabla_{1} &= \frac{6 \cdot 365 \times 10^{3}}{d^{2}} + \frac{1}{2} \times \frac{6 \cdot 365 \times 10^{3}}{d^{2}} \sqrt{4+4} \\
\nabla_{1} &= \frac{6 \cdot 365 \times 10^{3}}{d^{2}} \left[ 1 + \left[ \frac{1}{2} \times \sqrt{2} \right] \right] \\
\nabla_{1} &= \frac{6 \cdot 365 \times 10^{3}}{d^{2}} \left[ \frac{4 \cdot 41}{4} \right] \\
\nabla_{2} &= \frac{1}{2} \left[ \frac{(\sigma_{2} + \sigma_{3})}{d^{2}} - \sqrt{(\sigma_{2} - \sigma_{4})^{2} + 4 \times \sigma_{2}^{2}} \right] \times \frac{27 \cdot 27}{d^{2}} \\
\nabla_{2} &= \frac{1}{2} \left[ \frac{(12 \cdot 73 \times 10^{3})}{d^{2}} + o \right] - \sqrt{(\frac{12 \cdot 73 \times 10^{3}}{d^{2}} - o)^{2}} + 4 \times \left( \frac{6 \cdot 365 \times 10^{3}}{d^{2}} \right) \times \frac{1}{2} \times \frac{6 \cdot 365 \times 10^{3}}{d^{2}} \times \frac{1}{2} \times \frac{6 \cdot 365 \times 10^{3}}{d^{2}} - \frac{1}{2} \times \frac{6 \cdot 365 \times 10^{3}}{d^{2}} \times \frac{1}{2} \times \frac{6 \cdot 365 \times 10^{3}}{d^{2}} \times \frac{1}{2} \times \frac{6 \cdot 365 \times 10^{3}}{d^{2}} \times \frac{1}{2} \times \frac{1}{$$

To or 
$$\sigma_{2}$$
 or  $\sigma_{3}$  =  $\sigma_{4}$ 

[which is Maximum]

 $\sigma_{1} = \sigma_{5}$ 
 $\sigma_{1} = \sigma_{5}$ 
 $\sigma_{2} = \sigma_{5}$ 
 $\sigma_{3} = \sigma_{5}$ 
 $\sigma_{4} = \sigma_{5}$ 
 $\sigma_{5} = \sigma_{5}$ 

(ii) Maximum shear she stress theory  $\sigma_{5} = \sigma_{5}$ 
 $\sigma_{7} = \sigma_{2}$ 

(or)  $\sigma_{7} = \sigma_{5}$ 

(or)  $\sigma_{7} = \sigma_{5}$ 

which is maximum

 $\sigma_{7} = \sigma_{7} = \sigma_{7}$ 
 $\sigma_{7} = \sigma_{7} = \sigma_{7} = \sigma_{7}$ 
 $\sigma_{7} = \sigma_{7} =$ 

d= 13. 40 mm

3) Maximum Strain Theory 
$$(0.7)$$
  $(0.7)$   $(0.$ 

$$\frac{236.11 \times 10^{6}}{d^{\frac{1}{4}}} + \frac{6.806 \times 10^{6}}{d^{\frac{1}{4}}} + \frac{0.6 \left(\frac{15.366 \times 10^{3}}{d^{2}}\right) \left(\frac{-2.609 \times 10^{3}}{d^{2}}\right) = 11}{\frac{236.11 \times 10^{6}}{d^{\frac{1}{4}}}} + \frac{6.806 \times 10^{6}}{d^{\frac{1}{4}}} + \frac{21.05 \times 10^{6}}{d^{\frac{1}{4}}} = 100^{2}$$

$$\frac{1}{d^{\frac{1}{4}}} \left[ 236.11 \times 10^{6} + 6806 \times 10^{6} + 24.05 \times 10^{6} \right] = 100^{2}$$

$$\frac{1}{d^{\frac{1}{4}}} \times 266.96 \times 10^{6} = 100^{2}$$

$$\frac{266.94 \times 10^{6}}{100^{2}} = d^{\frac{1}{4}}$$

$$\frac{1}{d^{\frac{1}{4}}} \left[ 2.66 \times 10^{\frac{1}{4}} \times$$

I Distortion Energy Theory (von mises Hencky)

$$\frac{\sigma_{1}^{2} + \sigma_{2}^{2} + \sigma_{3}^{2} - \sigma_{1}\sigma_{2} - \sigma_{2}\sigma_{3} - \sigma_{3}\sigma_{1} - \sigma_{g}^{2}}{\left(\frac{15.366 \times 10^{3}}{d^{2}}\right)^{2} + \left(\frac{-2.609 \times 10^{3}}{d^{2}}\right)^{2} + o - \left(\frac{15.366 \times 10^{3}}{d^{2}}\right) \left(\frac{-2.609 \times 10^{3}}{d^{2}}\right) - o - o = 100^{2}}{\frac{238}{d^{2}}} = \frac{236.13 \times 10^{6}}{d^{2}} + \frac{6.806 \times 10^{6}}{d^{2}} + \frac{40.08 \times 10^{6}}{d^{2}} \Rightarrow 00^{2}$$

$$\frac{283.016 \times 10^{6}}{d^{\frac{1}{4}}} = 100^{2}$$

$$\frac{d^{\frac{1}{4}}}{283.016 \times 10^{\frac{1}{6}}} = d^{\frac{1}{4}}$$

$$100^{2}$$

$$d^{\frac{1}{4}} = 28301.6$$

$$d = [28301.6]^{\frac{1}{4}}$$
And  $d = 12.97 \text{ mm}$ 

A cylindrical shaft made of steel of yield strength

Too MPa 15 subjected to static Loads consisting of
bending moment loknom and a torsional moment 30km, m

Determine the diameter of the shaft using Two different

different theories of failure, and assuming a factor

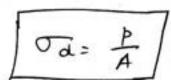
Of Safety of & Take E=210 GPA and Poisson's Ratio=0.25

DECCENTRY Loading - Direct and Bending Stress combined 00 An external Load, whose Line of action is paralle but does not coincide with the centroidal Axis of the machine components is known as eccentricload The distance blw the centroidal oxis of the machine component and the eccentric Load 15 calls eccentricity. Jet? 1. Direct Stress In the case shown it is d = compressi

Compressive Load

This is given by

Force Cross sectional Area



## 2. Bending Stress

This is due to bending

Moment (PXE). This results in different

types of Stresses on either side

of the natural axis of the bending

section, compressive on one-side

and tensile on the other.

Mb= Pxe

### troblem

1. A rectangular strut is 150 mm wide and 120 mm

thick. It carries a Load of 180 kN at an eccentricit

af lomm in a plane bisecting the thickness as shown

in Fig. Find the maximum and minimum intensities of

Stress v. 1180 kN

Stress in the Section.

## Given

wide b= 150mm

thick t= 120mm

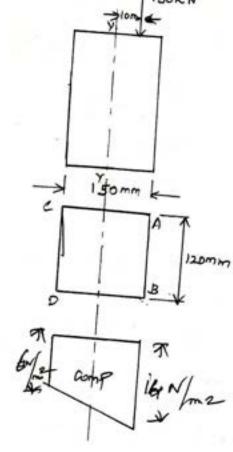
Load P = 180 KN = 180 X103 N

Eccentricity e= 10mm

## To Find

Maximum & Minimum Intensities of

Stress : o [combined stress]



## Soln

Maximum Intensities Stress OMax = Thing od + 06

Minimum Intersities Stress omin = 06-00 0d-06

Ta: direct stress due to compressive Load

Jb: bending stress due to eccentricity

Y= 1/2 = # b/2 = 150

Tyy= 163

y= 75 mm

 $T = \frac{4 \times b^3}{12} = \frac{120 \times 150^3}{12} = 33.75 \times 10^6 \text{ mm}^4$ 

Mb= Pxe = 180x103x 10 =1-8x106 Nomm

0b = 1.8 x 106 x 75

Jb= 4 N/mm2

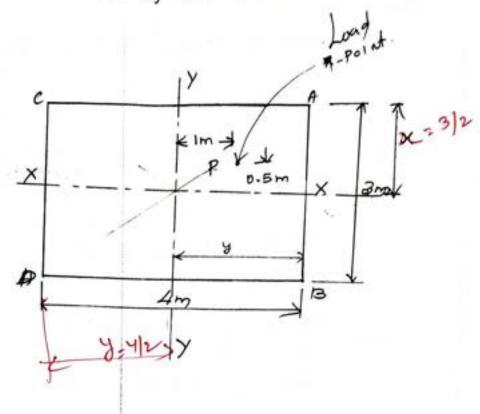
Omax = 0a+0b= 10+4

0 max = 14 N/mm2

Jmn = Jd \_ Jb = 10-4

0 min = 6 N/mm2

2) A masonry Pier of width 4m and thickness 3m, suf 3 a Load of 30KN as Shown in 71g. Find the Stresses developed at each corner of the pier.



Given

width b= 4m

thickness t = 2 m

Load P = 30KN = 30X103N

eccentricity at xaxis ex=0.5 m

enentricity at yours ey= 1m.

To Find

Stress at out corner (ABCD)

Mb Pxe

$$Txx = \frac{bt^3}{12} = \frac{4x3^3}{12} = 9m^4$$

$$T_{\gamma\gamma} = \frac{t b^3}{12} = \frac{3xH^3}{12} = 16m^4$$

Stress ad corner A
$$\sigma_{A} = \frac{P}{A} + \frac{P \times e_{x} \times x}{I_{xx}} + \frac{P \times e_{y} \times Y}{I_{yy}}$$

$$O_{H}^{2} = \frac{30}{12} + \frac{30 \times 0.5 \times 1.5}{9} + \frac{30 \times 1 \times 2}{16}$$

OB: 
$$\frac{P}{A} + \frac{P_{\times} e_{\times} \times \times}{T_{\times}} = \frac{P \cdot e_{y} \times y}{T_{yy}}$$

$$\overline{TB} = \frac{30}{12} + \frac{30 \times 0.5 \times 1.5}{9} = \frac{30 \times 1 \times 2}{16}$$

Stress at corner c
$$\nabla z = \frac{P}{A} - \frac{P \times x}{I \times x} + \frac{P \times e_y \times y}{I \times y}$$

$$20 \times 0.5 \times 1.5 \quad 30 \times 1.5$$

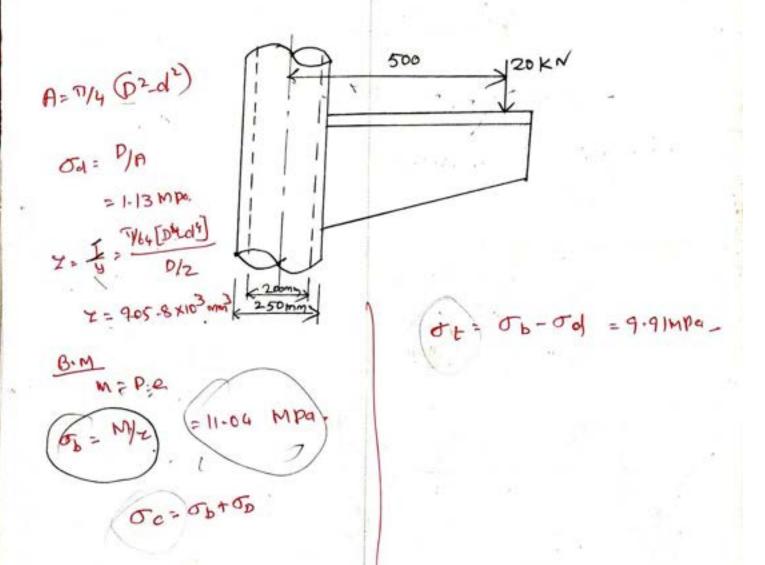
$$\overline{G} = \frac{30}{12} - \frac{30\times0.5\times1.5}{9} + \frac{30\times1\times2}{16}$$

## Stress at corner D

$$\overline{O} = \frac{P}{A} - \frac{P \cdot ex \times}{Txx} - \frac{P \times ey \times Y}{Tyy}$$

$$\sigma_{D} = \frac{30}{12} - \frac{30\times0.5\times1.5}{9} - \frac{30\times1\times2}{16}$$

3 A hollow circular column of external drameter 250 mm and internal drameter 200mm, carries a projecting bracket on which a load of 20 km rest, as shown; n fig. The leader of the Lead from the centre of column is 500 mm. Find the stresses and the side of the wlumn.



curved beams, the natural axis of the cross section is shifted towards the centre of the beam causing a non-Linear (hyperbolic) distribution of stress, as shown in Fig.

It may be noted that the newtral axis lies blw the Kentroidal axis and the Centre of curvature and always occurs within the curved beams.

Application: crane hooks

chain Links

Frames of punches

Presses

Planers etc.

Bending stress OB = Mb [Y] (or) ae[rn-y]

M: Bending moment

a: Area of cross section.

e= Distance from the centroidal axis to the Malural Axi

e= Y-Yn

m= Radius of wreature in natural axis

Y= DIStance of Fibre from NA.

Owler bending stress in Fibre

Thomax = Mbho

aero

Inside bediding stress in Fibre

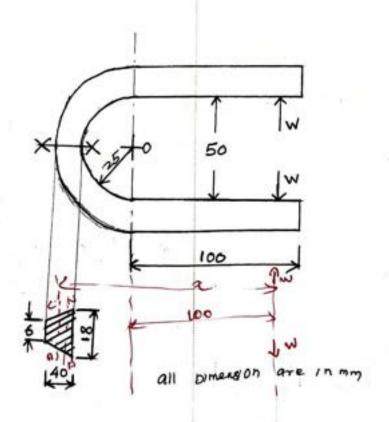
1 obly max = Mbhi aer;

Diagram Referrin David book Page No: 6.2

## Problem: 1

The Frame of a Punch Press is shown in Fig.

Find the stresses at the inner and outer surface at section X-X of the frame, if W = 50000.



Load w= 5000 N

To Find

Stress in inner and outer burface the Thimax & Tho

Soln

Maximum bending stress at the inner Gurface

[Ob] Max = Mbhi aeri

outer burface Maximum bending stress at the

Ob Max = Mbho

Mb = bending Moment at the centroidal Axis

W= 5000N Mb= Load x Distance = WX X

X = DIStantance blw the Load & controldal

From Dafabook X= 100 + R

46.3>

<6.3> R= Yi + h[bi+260] 3 [bi+60]

bi = 18 mm ] from Dayla 600/2 63

$$R = 25 + \frac{1200}{72} = 25 + 16.66$$

M&= WXX = 5000 X141.66

Mb = 708,300 N.mm

e- Distance blw centroidal axis and Mabural axis

e: R-Yn

$$\gamma_n = \frac{\frac{1}{2} \left[ b_i + b_0 \right] h}{\left( \frac{b_i \gamma_0 - b_0 \gamma_i}{h} \right) l_n \left( \frac{\gamma_0}{\gamma_i} \right) - \left( b_i - b_0 \right)}$$

The Sahon XX is subjected to Tensile lead of Wesooon and a Bending moment of M=708,354-mm WIT, Ot : N = 5000 10.42 P= 1/2 (18+6) 40 = 480mm2

Œ

$$\gamma_{n} = \frac{1}{2} \left[ \frac{18+6}{40} \right] 40$$

$$\left[ \frac{18\times65 - 6\times 35}{40} \right] \ln \left[ \frac{65}{85} \right] - \left[ 8-6 \right]$$

$$\Upsilon n = \frac{480}{24.35 - 12} = \frac{480}{12.352}$$

hi = Distance from the newtral Axis to the inner Surface

ho: Distance from the thetaineutral Asis to the outer surface

Stress in inner Surfue

Stross in outer burface

7:

Resultant offress on the inner ourface OR;

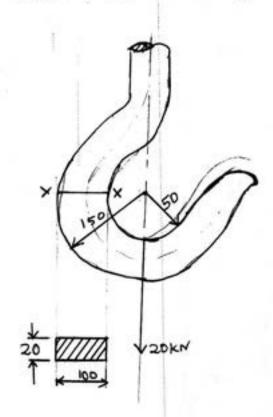
OR: = Of + Ob = 10.42 + 290.92 Max OR: = 301.94 N/mm²

Resultant Stress on the outer surface TRO

OR = 07 - 200.84 N/mm2

2. The Crane hook carries a Load of 20km as shown;

Fig. The Section at X-X 15 rectangular whose horizontal side 13 100mm. Find the stresses 9n the inner and outer fibres at the given section.



Load W = 20KN = 20X103N

Outer radius Yo = 150mm

Inner radius Yi = 50mm

### To Find

Stresses in Inner & outer surface Thimax &

O b Max.

30/2

Stresses in Inner Surface Topmax

blag = Mb hi

Stresses in owder surface To Max

Johns = Mo ho a er;

Mb= bending Moment

Mb = Load x Distance = WXZ

Load W = 20 x103 N

Distance  $x = R = \gamma_1 + \frac{h}{2}$ 

h = loomm

$$R = \gamma_1 + h/2$$
  
 $R = 50 + \frac{100}{2}$ 

Mb = WXR = 20x103x100

e: Distance blw centroidal axis and neutral axis.

rn: Radius of curvature of the neutral Axis

 $\gamma_{n} = \frac{h}{\ln \left(\frac{v_{0}}{v_{i}}\right)} \qquad h = 500 \text{ mm}$   $\gamma_{i} = 50 \text{ mm}$   $\gamma_{n} = \frac{100}{\ln \left(\frac{150}{50}\right)} \qquad \gamma_{0} = 150 \text{ mm (or) } \gamma_{i} + h$  = 50 + 10

1 = 91.02 mm

e= R\_Yn= 100-91.02

a: Area of cross section

a= = 100x 20

a= 2000 mm2

From DB6=

= 50+100=150m

hi= Distance from the neutral axis to the inside fibre

ho : Distance from the newtral axis to outside fibre

Maximum bending stress at the inside Fibre

Maximum bending stress at the outside 71 bre

Direct tensile stress of

Regultant stress in outside Surface TRO

### => Variable Stresses in Machine Parts:

The previous chapter, the Stresses due to Static Leading only.

But only a few machine parts are subjected to static Leading.

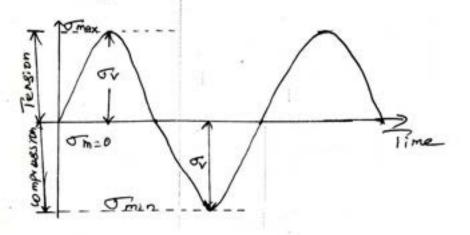
connecting rods, spring, pinion teath ele)

### Types of Varying Stress

- (i) completely reversed (or) cyclic stresses
- (ii) Fluctating Stresses.
- (17) Repeated stresses
  (14) Alternating stresses

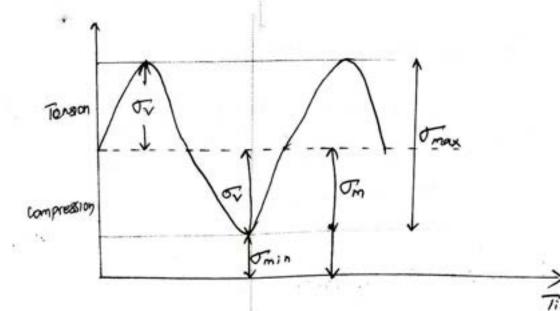
### (1) completely reversed (or) cyclic stresses:

Stresses which Change from one value of tension to the same Value of compression is known as completely reversed as cyclic stresses (TV)



(1) Fluctuating stresses

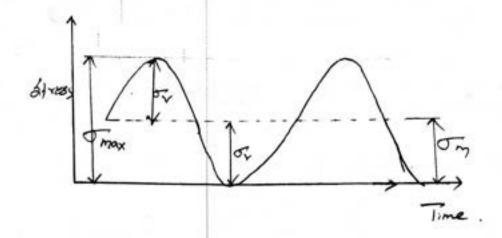
Stresses which vary from a minimum value to a maximum of same nature [compressive or tensile] are called as fluctuating stresses.



### (ii) Repeated Stress

This refers to a stress, which various drom

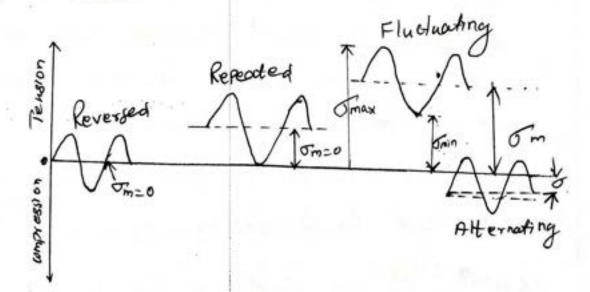
Zero to a maximum value of a same nacture.



### IV Alternative Stress

Stress varying from a minimum value to a maximum value of the opposite neture [from a minimum compressive to maximum tensile] is known as atternating st.

On common axis, these stresses may be plotted as given below.



Terms used in variable s	Stress conditions 27.67
1. Mean or average Stress	Om = Omax + Tmin
2. Variable Stress (or) amplit	tude Stress O a = O max - Umin
3. Stress Ratio R	
R= max	
It 13 completed	d reversed stress omin = - ma
R= Tripa = - Tripa	$\frac{c}{c} = -1$
4. Endurance Limit (5-1)  The endurence L	limit (5-1) of a material as
determined by the rotating	g beam method 10 Jur
reverged bending Load. members which are Subject	There are many mainine
reversed bending. Loads.	
Thus the end different for different	durance limit will also be types of loading
(1.42) O_1 = 0.50 4 700	steel [to bending]
U-1 = 0.450 4 1	for tension   compression]

= 0.30% for non services and alloyer

# > Factors affecting endurance strength:

1. Load Factor (KL)

This Factors varies with the loading type. reversed axial Load, reversed bendung etc.

Reversed bending

= 0.1 Reversed axial bending

= 0.6 reversed torsion

2. Surface finish factor (kst)

This factors arises due to the surface condition of the material. If the mirror surface is mirror like this factor equals unity. If poor surface is there, it is less than unity. The value of

kst = 0.9 for ground or cold rolled surface
= 0.7 to 0.85 machined Surface
= 0.3 to 0.7 hot rolled surface.

3. Size factor (KSZ)

A Large size specimen will have more defects
than a small one. so as size increase, this factor
reduce.

K5z = 1 d & 27.5 mm =0.85 d < 50 mm d> 7.5 mm 20.75 d > 50 mm

V CONTRACTOR

### 4. Reliability factor (KR)

Considerable scatter is found in fadigue street tests. A normal distribution gives a good agreement. With a standard deviation of 8% (0.08). This standard deviation can be subtracted from the 50% mean strength to obtain desired reliability.

KR = 1-0.080

Reliability Factor

survival rate	Deviation Multiplier, Df	Reliability factor Re
50	0	,
90	1.4	0.29
95	1.6	0.87
98	2.0	0.84
99	2.4	0.8
99.9	3.4	0.75

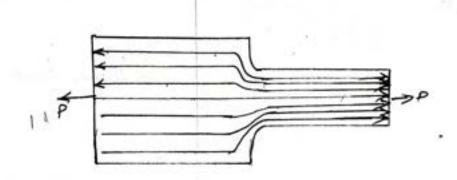
# 5. Miscellaneous factor k

- ature, impact factor may also be considered in determining the endurance limit

The actual or modified or component endurance strength

O-Im = O-1 KL KSZ &SF KRK

whenever a machine component change the shape of its cross section, the simple stress distribution no longer holds good and the neighbourhood of the discontinuity is different. This irregularity in the obtress distribution caused by abrupt changes of form is called stress concentration.



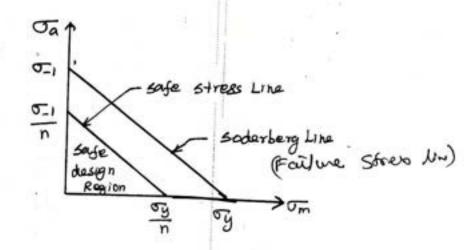
> Stress concentralition factor by 4007 47.87

Stress Concentration factor kt is defined as the ratio of the maximum stress at the change of Cross section to the normal stress.

Fatigue stress concentration factor ky < 7.6 >  $k_f = 1 + q (k_t - 1)$ Notch Sensitivity Jacker.  $q = \frac{k_f - 1}{k_t - 1}$ 

⇒ 50 deaberg and Goodman Diagrams

mediand is Duetile



In this piagram, amplitude stress and mean stress are plotted on y and x axes respectively.

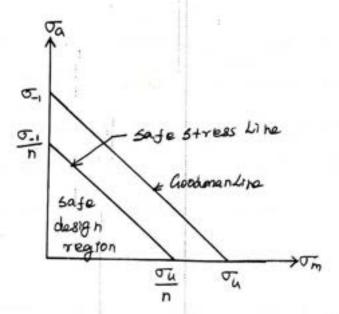
Soderberg Line Joins the endurance limit on y axis and yield stress on the x axis. This diagram a may be used for duetile Material.

$$\frac{1}{n} = \frac{\sigma_m}{\sigma_y} + \frac{\sigma_a}{\sigma_{-1}} - \frac{1}{n} = \frac{\tau_m}{\tau_y} + \frac{\tau_a}{\tau_{-1}} = \frac{\tau_{-1}}{\tau_{-1}} = \frac{\tau_{-1}}{\tau_{-1}} + \frac{\tau_a}{\tau_{-1}} = \frac{\tau_{-1}}{\tau_{-1}} + \frac{\tau_a}{\tau_{-1}} = \frac{\tau_{-1}}{\tau_{-1}} + \frac{\tau_a}{\tau_{-1}} = \frac{\tau_{-1}}{\tau_{-1}} = \frac{\tau_{-1}}{\tau_{-1}} + \frac{\tau_a}{\tau_{-1}} = \frac{\tau_{-1}}{\tau_{-1}} = \frac{\tau_{-1}}{\tau_{-1}}$$

Modelined Fan

$$\frac{1}{h} = \frac{\overline{\sigma_m}}{\overline{\sigma_g}} + k_g \frac{\overline{\sigma_a}}{(\overline{\sigma_{-1}}) k_L RSF RSZ RR R}$$

$$\frac{1}{h} = \frac{\overline{\tau_m}}{\overline{\tau_g}} + k_g \frac{\overline{\tau_a}}{(\overline{\tau_{-1}}) k_L RSF RSZ RR R}$$



If endurance strength (y oxis) is Jointed with Ultimal blress on the x oxis, it is called Goodman Life.

This diagram may be used for
$$\frac{1}{h} = \frac{\sigma_m}{\sigma_u} + \frac{\sigma_a}{\sigma_{-1}}, \quad \frac{1}{n} = \frac{\tau_m}{\tau_u} + \frac{\tau_a}{\tau_{-1}} \quad 27.47$$

Modified 
$$\perp = \kappa_t \left[ \frac{\sigma_m}{\sigma_u} + \frac{\sigma_a}{\sigma_{-1}\kappa_u} \right], \quad \frac{1}{n} = \kappa_t \left[ \frac{Z_m}{Z_u} + \frac{Z_a}{Z_{-1}\kappa_{all}} \right]$$

embined 
$$\frac{27.67}{\sigma_{eq}} = \frac{\sigma_W}{n} = \sigma_M + k_J \times \frac{\sigma_a \sigma_y}{\sigma_{-1} k_L k_{sf} k_{sz} k_R k_L}$$

$$\frac{\tau_{eq}}{\tau_{-1}} = \frac{\tau_{eq}}{\tau_{-1}} = \tau_{eq} + k_J \frac{\tau_a \tau_y}{\tau_{-1} k_L k_{sf} k_{sz} k_R k_L}$$

$$\frac{1}{\eta} = \left(\frac{\sigma_{eq}}{\sigma_y}\right)^2 + \left(\frac{\tau_{eq}}{\tau_y}\right)^2 \frac{\tau_{eq}}{\tau_{-1}}$$

Problem: based on varying stresses

1. A machine component is subjected to a flexural stress which fluctuates blw +300 MN/m² and -150 MN/m².

Determine the value of minimum ultimate strength according to 1. Gerber relation 2. Modified Good man relation, and 3. soderberg relation

Take yield strength = 0.55 Ultimate strength:

Endurance strength = 0.5 ultimate strength: and

Factor of safety = 2.

Given:

Maximum Stress  $\sigma_{max} = 300 \, \text{MN}/\text{m}^2 = 300 \, \text{X10}^6 \, \text{N}/\text{m}^2$   $\sigma_{max} = 300 \, \text{N}/\text{mm}^2$ Minimum stress  $\sigma_{min} = -150 \, \text{MN}/\text{m}^2 = -150 \, \text{N}/\text{mm}^2$ Yield strength  $\sigma_{ij} = 0.55 \, \sigma_{ij}$ endurance Strength  $\sigma_{-1} = 0.5 \, \sigma_{ij}$ Factor of safety  $\sigma_{-1} = 2$ 

To FINd

The Value of ultimate Strength ou by using 1) herber Relation

2). Modified sobderberg Good man relations

3) Sockerhere relation

### 1. Gerber Egn.

$$\frac{1}{n} = \left(\frac{\sigma_m}{\sigma_u}\right)^2 \times n + \frac{\sigma_a}{\sigma_{-1}}$$

$$\frac{1}{2} = \frac{75^2}{5u^2} \times 2 + \frac{225}{6.55u}$$

$$\frac{1}{2} = \frac{75^2 \times 2}{5 \cdot 2} + \frac{450}{5 \cdot 4}$$

$$\frac{1}{2} = \frac{11250}{\sigma_u^2} + \frac{450}{\sigma_u}$$

## 3. soderberg Relation:

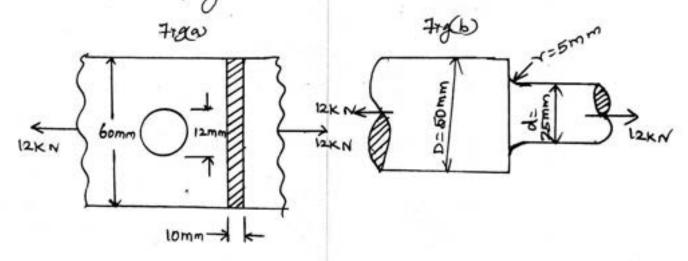
$$\frac{1}{2} = \frac{75}{0.5564} + \frac{225}{0.5564}$$

$$\frac{1}{2} = \frac{1}{54} \left[ \frac{75}{0.55} + \frac{225}{0.5} \right]$$

$$\frac{1}{2} = \frac{586.36}{54}$$

AND THE MONTH OF PROPERTY

- 2. Find the Maximum oftress induced in the following Courses taking stress concentraction into account.
  - 1. A rectangular plate 60mm x 10mm with a hole 12 mm chameter as shown in 74g.(a) and subjected to a tensile Load of 12km.
  - 2. A stepped shaft as shown in Fig(b) and carrying a tensile load of 12km.



Given

Rectangle place: width b=60 b(or) w=60 mm throk ness hept = 12 mm diameter of hole(a) = 12 mm Load P(or) W = 12 kN = 12 x10 N

### Stappedbar:

Padius of Hiller rasmm P = 12KN = 12X18 N

Maximum Strees o max

Soln

1. Redangle Place:

From Dada book P. NO<7.10>

$$\frac{P}{N_{om}} = \frac{P}{(W-a)h}$$

$$\frac{\sigma_{\text{hom}} = \frac{12 \times 10^3}{\left[60 - 12\right] 10}$$

Refer Diagram

Area Area

P= 12×103 N

h = 10 mm

W= 60mm

at the same

a = dra at dole

a= 12 mm

kt Refer Diagram 2087.10>

$$9|\omega = \frac{12}{60} = 0.2$$

### 2. Stepped bar.

$$\sqrt{12x10^3}$$
 $\sqrt{1/4x} = \frac{12x10^3}{25^2}$ 

3. A bar of circular cross section 15 subjected to alternating to tensile force varying from a minimum of 200KN to a maximum of 500KN. It is to be tensile manufactured of a material with an ultimate strengt of 900 MRADO and a endurance Limit of 700 MPa.

Determine the diameter of bar using of sectety factors related of 3.5 related to ultimate tensile strength and 4 related to endurance limit and a stress concentration. of 6.65 for fatigue Lpad. Use Goodman stright Line as tagging for design.

### Given data:

minimum Tensile Load Wmin = 200KN = 200X103N

maximum Tensile Load Wmax = 500KN = 500X103N

Ultimad tensile Strength ou = 900Mpg = 900 N/mm

endurance Limit on = 700 Mpa = 700 N/mm

Factor of safety for whinate strength

To = 3.5

For for ordunance limit no= = 4

Fatigue Stress concentration Factor K = 1.65

To Find:

Diameter of circular section 'd'

Soln

$$\frac{1}{n} = \kappa_{\frac{1}{2}} \left[ \frac{\sigma_{m}}{\sigma_{u}} + \frac{\sigma_{a}}{\sigma_{-1}} \right]$$

L7.67

$$\sqrt{\min_{n \in \mathbb{N}}} = \frac{W_{\min}}{Area} = \frac{200 \times 10^{3}}{\sqrt{1/4} \times d^{2}} = \frac{200 \times 10^{3}}{0.785 \times d^{2}}$$

$$\sigma_{m} = \frac{500 \times 10^{3}}{0.785 d^{2}} + \frac{200 \times 10^{3}}{0.785 d^{2}}$$

Streets Construits 
$$9 = 1 + 9 [Kt-1]$$

Notch sensitivity  $9 = 1 + 9 [Kt-1]$ 

Thursday construits  $9 = 1 + 1 [Kt-1]$ 

$$\frac{1}{n_{u}} : k_{\pm} \left[ \frac{\sigma_{m}}{\sigma_{u}} + \frac{\sigma_{a}}{\sigma_{-1}} \right]$$

$$\frac{1}{3.5} = 1.6 \left[ \frac{\frac{445.85 \times 10^{3}}{d^{2}}}{\frac{257.14}{257.14}} + \frac{\frac{1968.10^{3}}{d^{2}}}{\frac{175}{d^{2}}} \right]$$

$$\frac{1}{3.5} = 1.6 \left[ \frac{\frac{1445.85 \times 10^{3}}{257.14}}{\frac{175}{d^{2}}} + \frac{191.08 \times 10^{3}}{175 d^{2}} \right]$$

$$\frac{1}{3.5} = 1.6 \left[ \frac{1733.88}{d^{2}} + \frac{1091.88}{d^{2}} \right]$$

$$\frac{1}{3.5} = \frac{1.6}{d^{2}} \times \left[ 1733.88 + 1091.88 \right]$$

$$\frac{1}{3.5} = \frac{1.6}{d^{2}} \times 2825.76$$

$$\frac{1}{3.5} = \frac{4521.216}{d^{2}}$$

$$\frac{1}{3.5} = \frac{4521.216}{3.5} = 1291.77$$

$$\frac{1}{3.5} = \frac{4521.216}{3.5} = 1291.77$$

$$\frac{1}{3.5} = \frac{4521.216}{3.5} = 1291.77$$

d= 36 mm

A circular bar of 500mm length 15 supported (4)

freely at its two ends. It is acted upon by a

central concentrated cyclic Load having a minimu

Value of 20kN and a maximum value of 50kN.

Determine the diameter of bor by taking a factor of safety of 1.5, size effect of 0.85, burface finis.

If factor of 0.9. The material properties of bor are given by: ultimate strength of 650Mpa, yield strength of 500Mpa and endurence strength of 250Mpa.

Length 2= 500mm

Minimum Load Wmin=20×10<sup>3</sup>N

Maximum Load Wmax = 50×10<sup>3</sup>N

Factor of safety n = 1.5

Size effect factor  $k_{sz} = 0.85$ Burface Finish factor  $k_{sy} = 0.9$ Ultimate Strength  $\sigma_{u} = 650 \, \text{Mpa} = 650 \, \text{N/mm}^2$ Yield Strength  $\sigma_{y} = 600 \, \text{Mpa} = 500 \, \text{N/mm}^2$ Endurance Strength  $\sigma_{zy} = 350 \, \text{Mpa} = 350 \, \text{N/mm}^2$ 

To Find:

Diameter of shaft d.

Soln

By soderberg Eqn 
$$< 7.6 >$$

$$\frac{1}{h} = \frac{\sigma_m}{\sigma_y} + K_f \times \frac{\sigma_a}{\sigma_{-1} \times K_{SS} \times K_{SF}}$$

$$0 = 1.5$$
 $0 = 500 \,\text{N}/\text{mm}^2$ 
 $0 = 350 \,\text{N}/\text{mm}^2$ 

$$\frac{M_1}{T} = \frac{\sigma}{y} \qquad 27.10$$

Mb = bending Moment = Loadx Dis Mbmax = Maximum Load x Distance. Mbmax = 50×100 × 500 Mbmax = 6.25 x106 N. mm

Mbm) n = 2.5 x10 6 N.mm

$$\frac{63.66 \times 10^6}{d^3} + \frac{25.46 \times 10^6}{d^3}$$

$$\frac{\sqrt{m} = \frac{44.56 \times 10^6}{d^3}$$

$$Y = d_{12}$$
 $T = \sqrt[4]{64} \times d^{4}$ 
 $T_{1/4} = Z$ 

$$d^{3} = 241.88 \times 10^{3}$$

$$d = \left[ 241.88 \times 10^{3} \right]^{\frac{1}{3}}$$

$$d = 62.3 \text{ mm}$$

$$\frac{1}{1.5} = 1 \times \left[ \begin{array}{c} \frac{44.56 \times 10^6}{d^3} + \frac{19.1 \times 10^6}{d^3} \\ 650 \end{array} \right] \times \left[ \begin{array}{c} \frac{49.1 \times 10^6}{d^3} \\ 350 \times 0.85 \times 0.89 \end{array} \right] \times \left[ \begin{array}{c} \frac{44.56 \times 10^6}{d^3} \\ \end{array} \right]$$

$$\frac{1}{1.5} = \frac{68.55 \times 10^3}{d^3} + \frac{19.1 \times 10^3}{d^3}$$

$$\frac{1}{1.5} = \frac{140.68 \times 10^3}{d^3}$$

A 500mm diameter shaft is made from Carbon steel having Ultimate tensile strength of \$50 MPa. It is subjected to a torque which fluctuates blw 2000 Nom to - 800 Nom. Using soder berg method, Calculate the factor of safety. Assume builable Value for any other data needed.

### Given:

Moterial: Carbon steel

Maximum Torque Tmax = 2000N·m = 2002103 N·mm

minimum Torque Tmin = - 800 N·m = -800 x 103 N·mm

Dia meter orshard 2 500 mm

To Find

Factor of safety (n)

soln

doder berg Eqn 
$$27.17$$

$$\frac{1}{h} = \frac{Tm}{Ty} + k_f \frac{Ta}{T_{-1}x}$$

$$k_{52} \times k_{5p}$$

Zm = Zmax + Zmin

Zmax -By waing Torsion Egn 27.17  $T = \frac{M_t \times Y}{J}$   $M_t = T$   $Y = \frac{d}{2}$ J= 1/32 x d 4 Zmax = 200x103 x d/2 Tmax = 208x10 3 x 4 Zmax= 2000003 5003 Zmax = 1.018 x18 N/m/n2 Zmax = 0.0186 N/mm2  $Z_{min} = \frac{T_{min} \times \gamma}{3} = \frac{-800 \times 10^3 \times d_{\chi}}{T_{\chi_4}^7 \times d_{\chi_5}^7} = \frac{-800 \times 10^3}{T_{\chi_6}^7 \times 600^3}$ Zmin = - 4.01 x106 N/mm2 7min = -0.0325N/mm2

DB Endurance Simit Stress T-1 = 0.22 04 L1.42)

Ta: 0.0 285 N/mm2

47.67

$$\frac{1}{h} = \frac{7m}{7y} + k_f \frac{7a}{7-1 \, \text{ksz} \times \text{ksf}}$$

$$\frac{1}{h} = \frac{-6.95 \times 10^3}{180} + 1 \times \frac{0.0285}{44 \times 0.85 \times 0.5}$$

$$\frac{1}{h} = -6.72 \times 10^5 + 6.23 \times 10^4$$

$$\frac{1}{h} = 5.56 \times 10^4$$

$$\frac{1}{h} = 7.44 \times 10^4$$

$$\frac{1}{h} = 7.44 \times 10^4$$

$$\frac{1}{100} = h$$

h= 1843

A carditever beam made of cold drawn carbon steel of circular cross-section as shown in Fig. 15 Subjected to a Load which varies from -F to 3F. Determine the maximum load that this member can with stand for an indefinite life using factor. Of safety as 2. The theoretical stress concederation factor is 1.42 and the notch sensitivity is org. Assume the Following values

Ultimate Stress = 550 Mpg

Yield stress = 470 Mpg

Endurance Limit = 275 Mpg

Size factor = 0.85

Surface Sinish factor = 0.89

150

125

125

Civen:

Minimum Load  $W_{min} = -F$ Maximum Load  $W_{max} = 3F$ Fos n = 2

stoss Consentr Kt = 1.42
whon factor not sensitivity 9=0.9

Ju = 550 N/mm2

Yield stress by = 470 N/mm² Endurance Stress S-1 = 275 N/mm² Size fautor Ksz = 0.35 Surface finish fautor Kg = 0.89 dia of smallend d = 13mm dia of large end D = 20 mm Length of beam L = 150mm. Length of beam L = 150mm.

To Find:

Maximum Load 'F'

doln

Souder

Soderberg Eqn

1 = 0m + kg 0=, xksxx ksf 27.67

Im: mean stress

Om= Omax + Omin 47.6>

Tmax= MAXY Z7.1>

Mbmax = Wmax X Length of Small Shaft

=) Load at end.

I = Moment of irentia

$$\frac{1}{10} = \frac{56.51 \, \text{N}}{579} + \frac{56}{57} + \frac{56}{$$

Goodman Eqn
$$\frac{1}{h} = k_{t} \left[ \frac{\sigma_{m}}{\sigma_{u}} + \frac{\sigma_{a}}{\sigma_{-1} \times k_{sz} \times k_{sp}} \right]$$

$$\frac{1}{2} = 1.42 \left[ \frac{0.58F}{550} + \frac{1.15F}{275 \times 0.85 \times 0.89} \right]$$

$$\frac{1}{2} = 1.497 \times 10^{-3} F + 7.84 \times 10^{-3} F$$

$$\frac{1}{2} = 9.34 \times 10^{-3} F$$

$$F = \frac{1}{2 \times 9.34 \times 10^{-3}}$$

$$|F = 53.49 N|$$

to an axial Load which various from 150 N (compress) to 450 N (tension) and also a transverse Load abits free end which varies from 80 N and up to 120 N down. The candilever is of circular cross section.

It was at is of diameter 2d for the first 50 mm and of diameter d for remaining length. Determine Its dramater taking a factor of safety of &.

Assume the following Data

Yield Stress = 330MPa

Endurance limit in reversed loading = 300 MPa

correction factor = 0.7 in reversed according

Stress Concentration Factor = 1.44 for bending

= 1.64 for actal Loading

Size Effect Factor = 0.85

Surface effect factor = 0.90

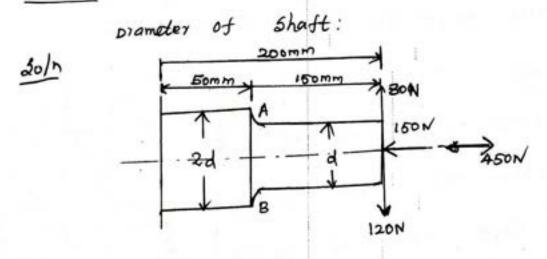
Notch Bensitivity Factor =0.90

#### Griven Data:

Length of cantilever l= 200mm
minimum Acial Load Wmin =-150 N
maximum Axial Load Wmax = 450N

Maximum Lansverge Load Wtmax = 120N Minimum transverse load Wt max = - 80 N Fos n= 2 yield stress oy = 330N/mm2 Endurance Strength 0-1 = 300 N/mm2 Correction factor Ka = 0.7 Kb=1 Stress Concentration factor kta = 1.64 ktb = 1.44 Size effect fautor Ksz = 0.85 burface effect factor ksf = 0.9 Notch sensitivity index = 9=0.9

To Find:



Total = Tega + Tegtrans

Teg= Equivalent Stress in axial Load at point A

$$\sqrt{\frac{381.96}{d^2}}$$
  $\sqrt{\frac{381.96}{d^2}}$   $\sqrt{\frac{190.98}{d^2}}$ 

$$\frac{\sqrt{min} = \frac{M_{bmin} \times y}{2} = \frac{-12 \times 10^{3} \times x}{2 \times 10^{3} \times x}$$

$$\frac{\sqrt{min} = -122 \cdot 23 \times 10^{3}}{12} = \frac{12 \times 10^{3} \times x}{12 \times 10^{3}}$$

$$\frac{\sqrt{min} = -122 \cdot 23 \times 10^{3}}{12} = \frac{12 \times 10^{3} \times x}{12}$$

Total equivalent stress at point A

$$\frac{\text{Total}}{\text{Fos}} = \frac{330}{2} = 165$$

$$\frac{1371.94}{211.67} = \frac{323.44}{2113.52 \times 10^3}$$

$$165 = \frac{911.07}{d^2} + \frac{323.44}{2113.52 \times 10^3}$$

By trial and error Method

de W. Sy mm

d= 12.73 mm

8 100. 48 Jud - 323 44 10 = 0

8. A hot rolled steel shaft is subjected to a torsional moment that various from 330 N·m clockwise and an applied bending Moment at a critical Section varies from 440 N·m to - 220 N·m. The Shaft is of uniform cross Section and no key way is presed at critical section. Determine the required shaft diameter. The material has an ultimate strength of 550 Mm MN/m² and a yield strength of HIDMN/m Strength, factor of safety of 2, size daylor of 0.85 and a surface finish factor of 0.85 and a surface finish factor of 0.62.

Tmax = 330 N.m (cw)
= 330x103N.mm

Tmin = - 110 N.m (ccw) =-110x103 N.mm

MbMax = 440 N.m = 440 X103 N.mm

Mbmin = -220 N.M = -220x103 Nomm

Th = 550 MN/m2 = 550 N/mm2

g = 410 MN/m2 = 410 N/mm2

0-1 = 04 = 550 2 = 2

0-1= 275 N/mm2

Fos n= 2 K5Z = 0.85 KS4 = 0.62

Drameter of Shaft. d

### Soln

Combined Stress

$$\frac{1}{h} = \left[ \left( \frac{\sqrt[3]{eq}}{\sqrt[3]{y}} \right)^2 + \left( \frac{\sqrt[3]{eq}}{\sqrt[3]{y}} \right)^2 \right]^{\frac{1}{2}}$$
  $\angle 7.67$ 

$$\frac{\sqrt{\frac{M_{bmin} \times y}{T}}}{\sqrt{\frac{T}{6u} \times d^{\frac{3}{2}}}} = \frac{-220 \times 163 \times x}{\sqrt{\frac{3}{2}}}$$

Z\_1 x KSzxKsf

$$T_{\text{max}} = \frac{T_{\text{max}} \times \gamma}{J} = \frac{330 \times 10^{3} \times \text{gt}}{T_{16}^{2} \times \text{d}^{3} \times \text{gt}}$$

$$\frac{1.68 \times 10^6}{d^3} + \frac{-860 \times 10^3}{d^3}$$

$$Z_{a} = \frac{Z_{max} - Z_{min}}{2} = \frac{\frac{1.68 \times 10^6}{d^3} - \frac{-.560 \times 10^3}{d^3}}{2}$$

$$\frac{V_{g}}{V_{g}} = \frac{\sigma_{g}}{2} \qquad 27.1$$

$$\frac{V_{g}}{V_{g}} = \frac{410}{2}$$

$$\frac{V_{g}}{V_{g}} =$$

$$\begin{bmatrix} \frac{1}{2} \end{bmatrix}^{2} = \left( \frac{10.62 \times 10^{6}}{410 \times d^{3}} \right)^{2} + \left( \frac{4.16 \times 10^{6}}{205 \times d^{3}} \right)$$

$$\frac{1}{4} = \frac{10.6}{400 \times d^{3}} + \left( \frac{20.29 \times 10^{3}}{43} \right)^{2}$$

$$\frac{1}{4} = \frac{670.81 \times 10^6}{d^6} + \frac{411.79 \times 10^6}{d^6}$$

9. A pulley is keyed to a shaft to a midway blw
Two bearings. The shaft is made of cold draw
steel for which the Ultimate strength is 550Mpg
and yield strength is 440 Mpa. The beat bendung
moment at the pulley varies from - 150 N. m to
trom as the torque on the shaft various

shaft for an indefinite life. The stress concentration factor for at the key way at the pulley in bending and in torsion are 1.6 and 1.3 respectively.

Take the following values.

Load correction factor = 1 in bending,

0.6 in torsion.

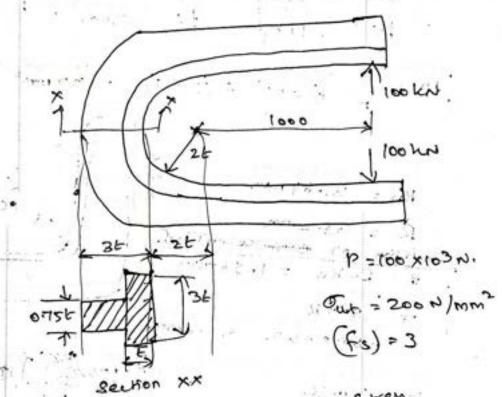
Size estent factor = 0.85.

H.w Refer Previous Problem.

Am d: 35mm

A.C. Clamp is subjected to a man. Local og'w' as shown in Fg. It the max tensile stress in the clamp is limited to 140 Mrs. King the Volue or load w. . Chiven: OF (non) = 140 N/mm2 Ri = 25mm; Ro : 25+252mm = 50 mm. bi = 19 mm; ti = 3mm; t= 3mm; h= 25m. A = (3 x 22) + (3 x 19) = 123 mm2 Rn = 31.64 mm. R = 33.2 mm. Colculation of our eccentricity (5) e= R-Rn = 1.56mm 2 = 50+R = 50+33 2 =83.2m M - W+ 2. = Wx 83.2mm Section - xx M= 83.20W N-mm adjulation of Orest terrice stens! -Direct Street, at section XX hi = 6.64 mm. max. Bending stress at Calculation of Benefity Street. OE = 0.008W N/mm² M P; = 0.112M N/WW5. WKT, calculation of ward 140 = Of + Obi. 140 = 0.008M +0.112M W= 1138N

The C- Frame of a book looked Capacity press as shown in 218. The material or the Frame is gray clifficon and the December of Sabety is 3. Determine the Dimensions of the Brame.



( Calculation of permissible Tensile Stress.

(ii) Calculation OA eccentricity (e) e = R - RN bi = 3t, h = 3t, ki = 2tRo = 5t, Ei = t t = 6.75t.

Rrv = 2.8134 +

R = 3t.

e= 0-1866E

#### UNIT- 2 I PART

## DESIGN OF COUPLING

#### Introduction:

shaft are usually available up to 7 metres Length due to inconvenience in transport. In order to have a greater length, it becomes necessary to Join two or more pieces of shaft by means of a coupling.

#### Purpose:

- > To Provide for the connection of shaft of unit that are manufactured separately such as a motor and generator and to provide for disconnection for reports or alternations. misallighment
  - > To provide for the misalinment of the shadt or introduce mechanical flexibility.
- > To reduce the Transmission of shock loads from one shaft to another.
- > To introduce protection against overloads.
- > It s should have no projecting parts.

- => Requirements of a Good Shatt coupling:
  - \* It should be easy to connect or disconnect.
  - \* It should transmit the full power from one shaft to the other shaft with out losses.
  - + It should hold the shafts in perfect abgnment.
  - \* It should reduce the transmission of short loads from one shart to another shaft.
  - + It should have no Projecting Parts.
- => Types of coupling:
  - H Thembl
  - 1. Rigid coupling

It is used to connect two shafts which are perfectly abgred. types of Rigid coupling a. sleeve en must coupling

- b. Clamp (or) split-muff or compression coupling. c. Flange coupling.
- 2. Flexible Coupling

It is used to connect two shatts both lateral and angular misallogment. types a. Bushed pin type coupling.

b. Universal Coupling

c. Old ham coupling.

It is simplest type of rigid coupling, made of cast iron. It consists of a hollow cylinder whose inner diameter is the same as that of the shatt.

It is fitted over the ends of the two shatt by means of a gib head kay.

The power is transmitted from one sharll to another a sharl by means of key and a sleave.

Diagram Reference < D.B XEROX 14>

besign Procedure for the muff coup

components of coupling [Must]

- 1. shaft
- 2. Sleeve or muft
- 3. Key.

#### Material ..

- 1. Shaff \_ castiron
- 2. Sleeve castiron
- 3. key castiron

# Design Procedure For Sleave or muff coupling

Stepi besign of shaft (d)  $T = \frac{7}{16} \times \left[ \frac{7}{9} \times d^{3} \right]$ 

T= + Tmax

Power P = 2TTNT No Tin N. m

Step-II

Dimension of mutt coupling. LD.B XEROX]

0 = outside dra of sleave = 2d+13

L= Length of Sleave L = 3.5d

Step-III Design of Sleeve, < DO XEROX 9>

Te = ?

Time [ Tan]

Design is sate

Step IX Design of Key < D BXEROX 9>

Dimension of key < 0.0 5.19>

to the hop ag .

t = thickness of key

Li length of key = 1

ah.

an failure of induced Shear.

T = l x W Tsk d/2

Ts = ?

75 L [75]

Design is safe.

b. failure of crushing stresses

T = lx[1/2] ock [d/2]

Oc= ?

0c 4 [04]

Design is safe.

Problems:

1. Design and make a next dimensioned sketch tof a muff coupling which is used to connect two steel shaft transmitting 40km at 350 Tpm. The material for the shaft and key is plain earbon steel for which allowable shear and crushing stresses may be taken as 40 Mpa and 80Mpa respectively. The material for the muff is costiron for which the allowable may be assumed as 15 Mpa.

#### Given data:

Power P = 40 KN = 400103 W

Speed N = 350 YPM

allowable others stress for short being

[Zs] = [Csk] = 40MPa = 40N/mm²

crushing stragginkey [OCK] = 80 MPa = 80 N/mm2

allowable shear streas for muff [Tcm] = 15 MPa: [Zcm] = 8000 15 15

To Find

Design the muff coupling

Soln.

perign of shaft dramater 'or'

T: 11/16 x [76] d3

Power P: 271 NT

T. Px60 = 40x103x60

T = 1100 N. M

T = 1100 x103 NIMM

1100×103 = 1/16× 40×d3

d = 52 mm say 1 55 mm

Step. B Dimension of Sleave Coupling L D. B XEROY>

D: out side dua of sleave = 2d+13 = 2x62+13

D= Immm

L= Longth of sleave = 3.5 d = 3.5 x 52

T= 182 mm

besign of sleave 2 Dis xEROXQ> Step-III

$$1100 \times 10^3 = \frac{\overline{u}}{16} \times 7_{cm} \left[ \frac{117^4 - 52^4}{117} \right]$$

Design 15 safe.

Step-IX Design of Key < D.B XEROX 97

Dimension of key 70 45.197

For shaft diameter 52 mm

Width b= 16 mm

thickness to lomm

Longth of key = = = 182 = 96 mm

1. Check For Induced Shear

Design 15 sate.

2. Check for crushing stress

[Och] = 80 N/mm2

l= 10mm

Design 15 Sade.

2. Design a mult coupling to connect two shaft transmitting HOKW at 1207pm. The permissible shear and crushing Stress for the shaft and key Maderial are 30 Mpa and 80 Mpa respectively. The material for mutt is cast iron with permissible shear strees. of 15 Mpg. Assume that the maximum Torque is 25% greater than the mean Torque. Given Data: Power P= 40KW = 40x103 W Spead M: 120 YPM Permissible Shear Stress For Shaft & stockey 9 [75] = [75k] = 30 MPA Permissible stress for key ze

[OCK] = 80 MP9 = 30 N/mm2

Permissible Shear Stress For mutt [Tim] = 15 Mpa

[TCM] = 15 N/mm2

besign a must coupling

Design of shaft diameter 'd' T = 7/16 x [Ts] x d3

T= 1.25 Tmin

Power P= 2TNTmin

40 x103 = 2x TIX 120x Tmin

Tmin= 3.183×103 N.M

Tmin = 3. 183 x16 Nomm

T = 1.25 x Tmin = 1.25 x 3. 183 x 106

T= 3.97 x106 N. mm

3.97 x106 = T/16 x 30x d3

d= 8716 mm say

Dimension of Loughing Sleave coupling Step-II

D= Diameter of 6/eeve = 2d+13 = 2x88+13

D= 189 mm

L= Length of sleeve = 3.5xd= 3.5x 88

1= 808 mm

Design of Sleeve < D.B XERUX 90 T= 71/16 x 7cm [ D4-24] ₹ 3.97 ×106 = T/16 × 7cm [1894-884] 7cm = 3.15 N/mm2 Lem show [Zen] = 30 + 15 N/mm2 Tem 4 [Tem] Design is sade. Step-IV besign of Key Dimension of key 6.19 > For Shaft diameter 52 88 mm width of key - 25 mm thickness of key - 14 mm Langth of key  $l = \frac{L}{2} = \frac{308}{2} = 154mm$ 1) check for Induced Shear T= lxwx Tsx x d/2 3. 97×106 = 154 × 25 × Tsk × 88 Tsk= 23.43 N/mm2 [Tsk] = 50 N/mm2 TSK < [TSK] Dougn 15 sade

6

(1) Check For induced crushing stress  $T = l \times t/2 \times \sigma_{ck} \times d/2$ 3. 97 ×106 = 154 × 14 ×  $\sigma_{ck} \times 88/2$ 

. JEK = 83.69 N/mm2

[DCK] = 80 N/mm

Design is not safe

Safer value increase in length of key

l= 170 mm

3.97 ×106 = 170 × 14 × JEK × 88/2

TCK = 76.82 N/mm2

Design is safe.

3) Design a muft coupling to connect two steel shafts transmitting 205kw power at 360 rpm. The shaft and key are made of plain courbon steel 30(8[Syt=Syc=100 N/mm²]. The steave is made made of grey cast iron FG 200 (Sut=200 N/mm²). The factor of safety for shaft and key is 4. For sleave, the factor of safety is 6 based on ultimate strength.

Given Power = 25 KW N=360 rpm TCM = 200 N/mm²

[Ts]= [Tsk] = 400 N/mm² \$ OCK = 400 N/mm²

## =) Design of Flange coupling

I Flange coupling usually applies to a coupling have two separate constitution flange.

Each Flange 15 mounted on the shaft and keyed to it. The faces are turned up at right angle to the Axis of the shaft.

The Two flanges are coupled together by means of bolt and Nuts.

Major	Parts	of	Flange	coupling
		*	process and forms	

1. Flange 2. Hub 3. shaft

4. bolt 5. Key

Material for Flange coupling

Shart 7 Bolt 9 steel. Keyy J

Hub y castiron.

-> Design Procedure For Un protected type Flange coupling or Rigid type Flange coupling: StepT Design of shaft dramater d' T= 11/16 x [75] an xd3 Power P= 2TT N Thean
60x1000 · Timean= ? Tmean = T. If service factor 13 given To S.Fx Thean T or Tmax = over Torque x Tmean. [ns] an = allowable shear stress for shaft material. 5+ep-2 Dimension: of Coupling outsided (xerex pg No: 9)
D= Diameter of hub= 2d Di= Dra of bolt circle = 3d Dz = outside dia of Flange = 4d

L= Length of hub = 1.5d tf: thrck news of Flange dr= n= No of bolt. 5tep-3: Design of bolt

a. Dra of both:

T= n x 1/4 x d12 x [7b] a11 x [1/2]

di=740 drameter at bolt 71nd.

b: check for crushing stress

T= nxdix to x ocb [Dil2]

OZb=7

JED Y [050]

Design is safe.

Step-4 Design of Key.

For key Dimension LPSG D.B 5.16>

l= Langth at key = l= L

w= width at key

t= thickness at key

a. Check for Induced Shear Stress

T= lxwx Zxxd12

tk= ? tk=?

TK L [ZK]all Design 15 safe. TK 15 Not Safe

Safer value Increape the Hength of Keyl

b. Check for crushing stress. T= lx(t/2) x Jc x (d12) oc is Not sade oc < [00] all Safer value increase Design 15 safe. the Longth at Key step-5 besign at hub. T= 7/16 x Zc x [ D7-d4] 74 = ? To < [ Te] all Design 13 sade. Design of Flange T= TX D2 X Zc x ty To Z[[a] Design 15 sade. value collection for [Ts] all [Tb] value collection for and [Tk] all, [OF] [Ock] [Och] all [Zc] a11 [ ts] = [ ] | n | n = 2 for s + end. [Tc] a11 = 04 = 04  $\left[\sigma_{3}\right] = \frac{\sigma_{3}}{4} = \frac{\sigma_{3}}{2}$ n= 9 Forci By= yield stress <PS4DB1.9> Oy= 36 regf/mm2 = 360 N/mm2 Ju: 220 N/mm2

Problem based on Rigid coupling (or) Unprotective type.

Flange

1. Design a rigid type of coupling to tonnect two shafts.

The Input Shaft transmits 37.5 km power at 180 rpm.

to the output Shaft through the coupling. The service factor for the application is 1.5. select switable material for various parts of the coupling.

Given

Power P= 37.5 kw = 37.5 x103 w

Spead N = 180 rpm

Service Factor = 1.5

Find:

Design the Rigid type coupling

soln

Step-1 prameter of shaft. d'

$$P = \frac{277 \, \text{NTesm}}{60 \times 1000} \Rightarrow 37.5 \times 10^3 = \frac{2 \times 11 \times 180 \times 7}{60 \times 1000}$$

THE # . 989 X106 N.mm

FA.

T= SF x 1.989 x106 = 1.5 x 1.989 x106

[Zs]a11 = ?

Assume Shaft, key, bolt material sted.

carboh steel 11.00

Ou= 63=7\$ Kgf/mm2

The 70 kgf | mm2 = 70x10

Ou: 700 N/mm2

Oy = 36 kg+ lam2 = 36x10

Oy: 360 N/mm2

[75] = [03]a11

[Zs] = 🖼 a11

h=2 For stoll. [0] = 0y = 0y

[0] = 360 = 180 N/mm2

[78] = 180 = 90 N/mm2

[25] an = [26] an = [26] = 90 N/mm2

[05]a11 = [05]a11 = [05]a11 = 180N/mm2

$$2.98 \times 10^{6} = \frac{11}{16 \times 90 \times d^{3}}$$

$$d^{3} = \frac{16 \times 2.98 \times 10^{6}}{11 \times 90}$$

d= 55.24 mm

Standard dramates d= 55 mm 27.20>

Step-2 bimension of coupling.

d= bia of shad = 55mm

D= out side dia of hub= 2d= 2x55= 110 mm

D= Dia of bolt circle = 3d= 3x65 = 165 mm

D= Dia of bolt circle = 3d= 3x65 = 165 mm

D= Dia of bolt circle = 3d= 3x65 = 165 mm

D= Dia of bolt circle = 3d= 3x65 = 165 mm

D= Dia of bolt circle = 3d= 3x65 = 165 mm

D= Dia of bolts = 4 lange = 4d= 4x55 = 220mm

L= Length of Flange = 4d= 4x55 = 27.5 mm

L= Length of Flange = 1.5d = 1.5x55 = 82.5

n= No of bolts = 4 Nos.

Step-3 Design of bolt  $T = n \times \sqrt[4]{4} \times d_1^2 \times \sqrt{2} b_{an} \times \frac{b_1}{2}$   $2.98 \times 16^6 = 4 \times \sqrt[4]{4} \times d_1^2 \times 90 \times \frac{165}{2}$   $d_1 = 11.3 \text{ mm}$ 

check for an crushing 61+285.

T= nxdixtfx Tcb x D1

2.98×106 = 4× 11.3× 27.5 × 0cb × 2

Ocb = 29.05 N/mm2

611 180 N MWZ

asp < [asp] 411

Design is safe.

Step- 4 besign of key.

For key bimension < PSG D.B 5.16>

For shaft diad: 55 mm

W= width at Key= 16mm

t = Thick nees of key = 10mm

l= Length at Key - l= L= 82.5 mm.

a. Cheek for Induced Shear Stress

Design is safe.

1. Check for crushing stress

Design is Not safe

So safer value increase the Langth of key

Step-5 Design of hub

$$T = \frac{17}{16} \times Z_{c} \times \left[ \frac{D^{4} - d^{4}}{D} \right]$$
 $2.98 \times 10^{6} = \frac{17}{16} \times Z_{c} \left[ \frac{110^{4} - 55^{4}}{110} \right]$ 
 $\left[ \frac{7}{16} \right] = \frac{7}{16} \times \frac{7}{16} \times \frac{7}{16} = \frac{110^{4} - 55^{4}}{110} = \frac{7}{16}$ 

For hub & Flange Material is castiron
$$\begin{bmatrix} \left[ \mathcal{T}_{c} \right]_{a_{11}} = \frac{\sigma_{u}}{n} = \frac{\sigma_{u}}{q} & n = 70s \\ \text{LPSGDB} \cdot 1.4 > & n = 9 \left[ 707 \text{ Ce} \right] \end{bmatrix}$$

CI I 20 BAE Tu= 220 N/mm2

$$\left[ \eta_{c} \right]_{an} = \frac{220}{9}$$

Design 15 Safe.

$$T = \pi \times \frac{D^2}{2} \times 7c \times t_f$$

Design 15 safe.

2. Design a cast 1000 flange coupling for a mild solve steel shaft dransmitting 90kW at 2500pm, the allowable shear stress on the shaft makes as a solid is 40 Mpa and Angle of twist is not to exceed 1 in a length of zometers. The allowable shear stress in the coupling bolt is 30 Mpa. Take G = 84KN mm2.

Given.

Speed N= 250rpm

allowable Shear [Zs] = 40MPa = 40xx N/mm² stress for shaft

Power P= 90KW =90x103W

Angle of twist 0=1° = 11/180 = 0.0175 radray

Allowable shear stress for bolt [Tb] = 30MPA=30N/mm

Rigidity Modulus Ki: 84 KN/mm2 = 84x13 N/mm2

To Find:

Dreign a CI Flange coupling.

6tep-I besign of shaft diameter d' soln T= T/16 x d3x [75] Power P= 2TT N Timean
60 × 1000 90×103 = 2×11 × 250× Times Tmean = 3440 x18 Nomm Thean = T. = 3440 X103 Nomm T= 3.44x106 Nomm 3 3 -44 × 106 = TT × d3 × B = 40 d3 = 16 x 3.44 x 106 d=75.94 mm Standard dra [d = 80 mm] LPS4 0-B7-20>

Considering Rigidity of Shaft
$$\frac{T}{J} = \frac{G_1 B}{L}$$

$$\frac{3.044 \times 10^6}{T_{32}^2 \times d^4} = \frac{84 \times 10^3 \times 0.0175}{20}$$

HEP

$$\frac{3.44 \times 10^{6}}{0.0981 \times 73.5}$$

$$d^{4} = \frac{3.44 \times 10^{6}}{0.0981 \times 73.5}$$

$$d^{4} = 4.77 \times 10^{5}$$

$$d = [4.77 \times 10^{5}]^{1/4}$$

$$d = 26.28mm$$

Ta Standard dra d=

LPSG D-B7207

Taking Largest of the two values.

d= Bomm:

D= outside dua of hub= 2d= 2x80=160mm

D= Dra of bolt errele = 3d = 3x100= 240mm

D= Dra of outer Flange = 4d=4x80 = 320mm

L= Length of Flange = 1.5xd = 1.5x80=120mm

tf = thickness of Flange = d|2= 80 = 40mm

N= No of bolt= 4 Nos.

Btep-3 besign of boit

check for crushing stress.

e.

4

1

Step-4 Design of Key

key Dimension LP54 0.8.5.16>

For shaft dra d= 80mm

width of key W= 22mm

throkness of key t= 14mm

length at key = Length at flange = 120mm

l= 120mm

a. Check for Induced shear stress

T= lx wx TK xd/2

3.44 ×106 = 120× 22× TK× 80

TK = 32.57 N/mm2

[ZK] = 40 N/mm2

Design is safe.

b. cheek for crushing stress

T= lx t x JCKx d

3.44 ×106 = 120× 14 × OCK × 80

OCKJE 102.38 N/mm2

JEK > [JEK] all Design Is Not Safe.

Design is safe.

Design of hub
$$T = \sqrt{11/16} \times Zc \times \left[ \frac{D^4 - d^4}{D} \right]$$

For hub & Flange Material TS castiron.

$$\begin{bmatrix} \overline{Z} \cdot \overline{J} = 1 \\ \overline{Q} \end{bmatrix} = \begin{bmatrix} \overline{Q} \cdot \overline{Q} \\ \overline{Q} \end{bmatrix} = \begin{bmatrix} \overline{Q} \cdot \overline{Q} \\ \overline{Q} \end{bmatrix} = \begin{bmatrix} \overline{Q} \cdot \overline{Q} \\ \overline{Q} \cdot \overline{Q} \end{bmatrix}$$

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$$= \begin{bmatrix} \overline{$$

(I \$20 SAE OU= 220N/mm2 4 LPS4 D.B1.47

To 
$$< [Tc]_{a11}$$

Design 15 safe

Step 6 Design of Flange

 $T = IT \times \frac{D^2}{2} \times Tc \times tf$ 
 $3.44 \times 16^6 = TI \times \frac{160^2}{2} \times Ts \times 40$ 
 $Ts = 2.13 N |_{mm^2}$ 

75 L [75]a11

besign is safe.

The shafts are keyed to the Flange hub.

The permissible stresses are given below shear stress on shaft = 100 Mpa

Bearing (er) Crushing Stress on shaft = 250 Mpa

shearing stress on keys = 100 Mpa.

Bearing stress on keys = 250 Mpa.

Shearing stress on CI = 200 Mpa

Shearing stress on bolts = 100 Mpa

After designing the various Elements, make a neat sketch of the assembly indicating the important dimensions. The stresses developed in various members may be checked if thumb rules are using for fixing the dimensions.

## Criven data:

Torque  $T_mean = T = 250 \text{ Nom} = 250 \times 10^3 \text{ Nomm}$ No of bolt n = 4Permissible Stresses on Shaft  $[T_6]_{a11} = 100 \text{Mpa} = 100 \text{N/mm}^2$ Crushing Stress on shaft  $[T_6]_{a11} = 250 \text{Mpa} = 250 \text{N/mm}^2$ Shear stress on key  $[T_6]_{a11} = 100 \text{Mpa} = 100 \text{N/mm}^2$ Bearing Stress on key  $[T_6]_{a11} = 250 \text{N/mm}^2$ Shearing Stress on  $[T_6]_{a11} = 250 \text{N/mm}^2$ Shearing Stress on  $[T_6]_{a11} = 250 \text{N/mm}^2$ Shearing Stress on  $[T_6]_{a11} = 200 \text{Mpa} = 200 \text{N/mm}^2$ Shearing Stress on bolt  $[T_6]_{a11} = 100 \text{Mpa} = 100 \text{N/mm}^2$ 

To Find:

besign of Rigid Flange coupling.

soln

StepI Design of Shaft diameter 'd'

Say

Standard dra < PSG D.B.7.20>

d: 25mm

Step-2 Dimension of coupling.

D= oud side dra of hub= 2d= 2x25= 50mm

DI = Dia of bolt circle = 3d = 3x85= 75mm

D2 = Dia of ouder Flange = 4d= 4x25= 100mm

L= Langth of Flange = 1.5d= 1.5x25 = 37.5mm

tf = thrckness of Flange = \frac{d}{2} = \frac{25}{2} = 12.5mm

N= No of bolt = 4 Nos given.

Step.3 Design of bolt
$$T = n \times \sqrt[4]{4} \times di^{2} \times \left[ \frac{7}{6} \right]_{aii} \times \frac{D_{i}}{2}$$

$$\frac{250 \times 10^{3}}{4} = 4 \times \frac{17}{4} \times di^{2} \times 100 \times \frac{75}{2}$$

$$di = 4.6 \text{mm}$$

Check For crushing stress

T = n x di x tf x \(\tau\_{cb} \times \frac{b\_1}{2}\)

250 \(\times 10^3 = 4 \times 4.6 \times 12.5 \times \frac{b\_0}{2}\)

\[
\tag{7.5} \times \tag{7.5} \times \tag{7.5} \times \tag{7.5} \tag{7.5} \tag{7.5}

JCb < [Job]all Design is safe.

Step 4 besign of key

Key dimension LPSG D.B \$.16> Width of key w= 8mm

thick ness of key t= 7mm

Length of key 2=1=37.5mm

a. Check for Induced shear stress

b. Check for crushing stress.

Design 15 safe.

$$250\times10^3 = \frac{11}{16} \times 7_{CX} \left[ \frac{50^4 - 25^4}{50} \right]$$

Design is safe.

Design of Flange

$$T = \pi \times \frac{D^2}{a} \times Zc \times tf$$

$$\frac{250\times10^3}{10^3} = \pi \times \frac{50^2}{2} \times 70 \times 12.5$$

-> Protective type Flange coupling.

The small though change the \$700 Of Flange Part will be extent.

The find the throkness of Protecting Flange to in step-2.

## Problem:

Design and draw a protective type of cast tron flange coupling for a steal shaft transmitting 15 kw at 2007pm. and having an allowable shear Stress of 40 Mpa. The working strang in the bolts should not exceed 30 Mpa. Assume that the same material 15 used for shaft and key and that the crushing stress 15 Twice the value of shear stress. The Maximum Torque 15 25% greater then the full load Torque. The shear stress for castiral is 14 Mpa.

Gover P = 15 kw= 15x103 W Spead N= 200 rpm.

allowable shear stress for shaft [75]a11 = 40 MPa. = 40 N/mm2

in

6.

working stress in bottom = 30 MPa: 30 N/mmboln

Shear Stress in bott [76] = 30 N/mm²

Same Material Jay Shaft & Van

Same Material For Shaft & Key

[To]a = [Tk] = 40 N/mm²

Crushing stress = Twice the Shear stress  $[\sigma_{CK}] = [\sigma_{S}]_{ij} = 2 \times [z_{S}]_{a_{ij}} = 2 \times 40$   $[\sigma_{CK}]_{a_{ij}} = 80 \text{ N/mm}^2$ 

T=1,25 X Tmean.

Shear stress for CI = HMpa = 14 N/mm2

To Find:

Design & Protective type toupling Drawing

Soln step I Design of shaft drameter d'

T= 16 × [25] = 1 × d3

Power P= 2TTN Tmean
60 x1000

15x103 = 2x11x 200 x Timean
60 x1000

Tmesh= 7.16x105 N.mm

T= 1.25 x Tmean

T= 1.25 x 7.16x105

T= 895 X10 5 N.mm

8.95 x105 = TT x 40 x d3

d= 48.48 mm

Say

Standard Orameter a= 50mm < PSG 0-87.20)

Step-2 Dimension of coupling.

D= outside dra of hub = 2d = 2x50=100mm

DI = DIA of bolt circle = 3d = 3x50=150mm

Da = DIA of outer Flange = 4d = 50x4= 200mm

L= Length of Flange = 1.5d = 1.5x50 = 75mm

tf = throkness of Flange = d/2 = 50 = 25mm

tp = throkness af Protecting Flange = d/4=52=12.5r

h= No of bolt = 4NOS

Step 3 Design of bolts

$$T = n \times \frac{\pi}{4} \times d^2 \times \left[ \frac{7}{5} \right]_{all} \times \frac{D}{2}$$
 $T = n \times \frac{\pi}{4} \times d^2 \times 30 \times \frac{150}{2}$ 

8. 95×10<sup>5</sup> =  $4 \times \frac{\pi}{4} \times d^2 \times 30 \times \frac{150}{2}$ 

Check for erushing stress

 $T = n \times d_1 \times d_2 \times \sigma_{cb} \times \frac{D}{2}$ 

8.95 ×10<sup>5</sup> ×  $4 \times 11.25 \times 25 \times \sigma_{cb} \times \frac{160}{2}$ 
 $\sigma_{cb} = \frac{106N}{411} \times \frac{100}{2} \times \frac{100}{411} = \frac{100}{411} \times \frac{100}{411}$ 

Design is safe.

Step 4 Design of key

key Dimension LPSG D.B 5.16>

width of key w= 16 mm

thickness of key to 10mm.

Length of key l=L = 75mm

a. Check for induced Shear stress

T= lxw x Txxd

8.95 x10 = 75 x 16x 7x x 50

ZK= 29.83 N/mm2

[ZK] 911 - 40 N/mm2

Zr L [Zr]ail Design 15 safe.

b. Check for Induce crushing stress

JEKE 96.46 N/mm2

[Ock] all = 80 N/mm2

OEK > LOEKJall

beergn 13 not safe.



besign is safe.

Step-6 Design of Flange

> Design of Florible coupling:

A Flescible coupling used to so as the unduce he permit misalignment of the shaft without unduce absorption of the power which the shaft are transmitting.

Design Procedure for Flexible coupling (or) bushed Arn
Coupling.

Step I Design of shaft dramder d'

x=?

Thean = IN N.mm

T = Thean

It service Factor is given

T= 6.F x Timean.

step-2 Dimexalon of coupling

d = dra of shaft D = outer diameter of hub = 2d

Di= Pcd of bolt = 3d

02 = outer drameter of Flange = 4d

dn= no of bolt

 $d_1 = D_1 a$  of bolt =  $\frac{0.5d}{\sqrt{n}}$ 

L= Length of Flange = 1.5d

t = thickness of flange ty = d .

Step-3 sim Design of Key

Dimension of Key 2 P.54. D.8 5.19)

thickness of key t

length of key l= L

a. Cheek for Induced Shear

T= lxwxZk x d

ZK=?

ZK L[ZK]all

beergn 15 safe.

Design 15 sade.

Step- 4 Deangn of bollt

a. Check For Induced Sheat

$$T = n \times \frac{\pi}{4} \times d^2 \times Z_b \times \frac{D_1}{2}$$

besign is sade.

b. Maximum bending stress & shear stress

27.2 Nodifred)

$$Z = \frac{7}{32} \times d^{3}$$

$$\angle D.B \times 10 >$$

$$M = W \left[ \frac{1}{2} + 5 \right] \angle D.B \times 9 >$$

$$T = \frac{1}{4} W \left[ \frac{7}{6} \right]_{a_{11}} \times \frac{d}{2} \angle D.B \times 10 >$$

$$W = ?$$

$$\angle D.B \times 10 >$$

$$A = \frac{1}{4} W \left[ \frac{7}{6} \right]_{a_{11}} \times \frac{d}{2} \angle D.B \times 10 >$$

$$W = ?$$

$$A = \frac{1}{4} W \left[ \frac{7}{6} \right]_{a_{11}} \times \frac{d}{2} \angle D.B \times 10 >$$

$$W = ?$$

$$A = \frac{1}{4} W \left[ \frac{7}{6} \right]_{a_{11}} \times \frac{d}{2} \angle D.B \times 10 >$$

Step. 5 Design of #10 hub

$$T = \frac{\pi i}{16} \times 7c \times \left[ \frac{D^4 - d^4}{D} \right]$$

$$7c = ?$$

Design 15 Sate.

Zough is not given assume 2.

## Problem based on Flexible coupling:

- 1. Design a bushed -pin type of flexible coupling to connect a pump shaft to a motor shaft transmitting 32kw od 960rpm. The overall torque 15 20 percex more than the mean Torque. The material properties are as Follows.
  - (1) The allowable shear and crushing stress for shaft moderial is 40MPa and 80 MPa. respectively
  - (ii) The allowable Shear stress for Castiron 75
    15Mpa.
  - (ii) The allowable bearing Pressure for Mubber bush is 0.8 N/mm2
  - (IV) The Material of the PPn 18 same as that of shatt and key.

Draw nead sketch of the coupling.

## airen douta:

Bushed - Pin type

Power P = 32 KW = 32 XIO3 AND Speed N = 9607PM

T=20.Tmean = 1.2 Tmean

allowable shear stress for shaft, key, bolt or) pins

Crushing stress for shaft, key, bottor pins

The allowable shary stress on II =

Bearing Pressure Pb = 0.8 N/mm2

To F71nd

Design the Bush pin coupling.

soln.

Step-i Design of Shatt diameter d'

Thean = 3.18x105 Nimm

T = 1.2 x Threan = 1.2 x 3.18 x105

T= 3.81x105 N.mm

Standard digneter d= 40mm LPSG D.B 7.20>

Step-2 dimension of coupling

d= Dra as shaft = 40mm

D= order Drameter of hub = 2d= 2x40= 80mm.

DI= P.Cd of bolt = 3d= 3x40= 120mm

Dz= outer drameter of Flange = 4d = 4x40 = 160 mm

di= bra at bot = 0.5 d = 0.5 x 40 = 11.54mm

u= No of polt>

h= 3 NOS

L= Length of flange = 1.5d =  $45 \times 40 = 60$ mm t=1 hickness of flange  $t_f = \frac{d}{2} = \frac{40}{2} = 20$ mm.

6tep-3

besign of key 20.8 5.19>

Dimension of key W= 12mm

Width of key W= 12mm

Hickness of key t= 8mm

Length of key != 1.5 d= 1.5x40=60mm.

a. Check For Induced Shear

$$T = 1 \times W \times T \times \frac{d}{2}$$

$$3.81 \times 10^{5} = 60 \times 12 \times T \times \frac{d}{2}$$

$$Tk = 26.48 \text{ N/mm}^{2}$$

$$Tk = 26.48 \text{ N/mm}^{2}$$

$$Tk = 26.48 \text{ N/mm}^{2}$$

$$Tk = 10 \text{ N/mm}^{2}$$

$$Tk = 10 \text{ N/mm}^{2}$$

$$Tk = 10 \text{ N/mm}^{2}$$

$$T = 10 \times \frac{d}{2} \times T \times \frac{d}{2} \times T \times \frac{d}{2}$$

$$3.81 \times 10^{5} = 60 \times \frac{8}{2} \times T \times \frac{d}{2} \times \frac{d}{2}$$

$$3.81 \times 10^{5} = 60 \times \frac{8}{2} \times T \times \frac{d}{2} \times \frac{d}{2}$$

$$T = 10 \times \frac{d}{2} \times T \times \frac{d}{2} \times T \times \frac{d}{2}$$

$$T = 10 \times \frac{d}{2} \times T \times \frac{d}{2} \times T \times \frac{d}{2}$$

$$T = 10 \times \frac{d}{2} \times T \times \frac{d}{2} \times T \times \frac{d}{2}$$

$$T = 10 \times \frac{d}{2} \times T \times \frac{d}{2} \times T \times \frac{d}{2} \times T \times \frac{d}{2}$$

$$T = 10 \times \frac{d}{2} \times T \times \frac{d}{2} \times \frac{d}{2} \times T \times \frac{d}{2} \times$$

a. Check for induced shear

$$T = n \times \frac{\pi}{4} \times d_{1}^{2} \times T_{b} \times \frac{D_{1}}{2}$$

$$3.81 \times 10^{5} = 3 \times \frac{\pi}{4} \times 11.54 \times d_{1}^{2} \times \frac{D_{1}}{2}$$

$$T_{b} = 20.23 \text{ N/mm}^{2}$$

besign is safe.

b) Maximum bending stress & stream stress

· Of- beach

The bending stress

$$\frac{Mb}{2}$$
:  $\frac{\sigma_b}{\gamma}$   $\angle 7.1>$ 

400xas Wo Pd xdxxl

$$\begin{array}{lll}
\nabla_{b} M_{qx} &= & \frac{\nabla_{b}}{2} + \frac{1}{2} \int (\nabla_{b})^{2} + h \cdot \nabla^{2} \\
&= & \frac{1.8407}{2} + \frac{1}{2} \int \frac{1.8407}{(21-23)^{2}} + h \cdot (0.08q)^{2} \\
&= & \frac{1}{2} \int \frac{1.8431 \, N}{mm^{2}} \int \frac{1.8431 \, N}{mm^{2}} \cdot \frac{1.8431$$

Step-5 Design of Flange
$$T = \pi \times \frac{D^2}{2} \times 7c \times tf$$

$$3.81 \times 10^5 = \pi \times \frac{80^2}{2} \times 7c \times 20$$

$$T_c = 1.89 \text{ N/mm}^2$$

$$Z_c = 15 \text{ N/mm}^2$$

$$Z_c = \sqrt{2c} J_{all}$$
Design 15 Safe.

Step 6 Design of hub

$$T = \frac{\pi}{16} \times Z_{c} \times \frac{D^{4} - d^{\frac{1}{4}}}{D}$$

$$3.81 \times 10^{\frac{1}{4}} = \frac{\pi}{16} \times Z_{c} \times \frac{80^{\frac{1}{4}} - 10^{\frac{1}{4}}}{80}$$

$$Z_{c} = 4.04 \text{ N/mm}^{2}$$

$$Z_{c} = 15 \text{ N/mm}^{2}$$

$$Z_{c} < [Z_{c}]_{aii}$$

$$Design is sadc,$$

$$Step - 7 Design of bush < D.e. × 9.>$$

$$W = P_{b} \times d_{2} \times l$$

$$7.93 = 0.8 \times d_{2} \times 60$$

$$\boxed{d_{2} = 0.165 \text{ mm}}$$

$$\boxed{Dush} = \frac{W}{L \times d_{2}}$$

$$\boxed{Dush} = \frac{7.93}{60 \times 0.165}$$

$$\boxed{Dush = 0.8 \text{ N/mm}^{2}}$$

$$\boxed{Dush = 0.8 \text{ N/mm}^{2}}$$

Toush = 0.8 N/mm

Toush = 1 Tous Jan

2 = [ Tous Jan

2 Toush au = 4 N/mm

Obven Loomsing 11
Design is safe.

It Notgiven Zoughez Nfmn2-Cassume) connecting a motor shaft to a pump shaft for the following service conditions.

Power to be transmitted = HOKW

Speed of motor shaft = 1000 r.pm

diameter of motor shaft = 50mm

diameter of the pump shaft = 45mm

The bearing Pressure in the rubber bush

and the pins allowable stress in the pins

are to be timited to 0.45 N/mm² and 25 Mpa

respectively.

H.W

3. Design a flange coupling [bush type] to transmit the at 760 rpm with a service factor of 1.2 for 6haft, key, bolt with permissible stress 50 N/mm² for castiron shear stress 15 N/mm² for the stress for bush 15 2 N/mm² for Key trushing stress of 100 N/mm².

# TEMPORARY AND PERMANENT JOINT

Threaded fasters - Bolted Joints - Including

deceptive loading - Knuckle Joints - cotter Joints 
Welded Joints , riveted Joints For Structures 
theory of bonded joints.

= Design of welded Joints: <PSG DB 11.1 - 11.6>

#### Introduction:

A welded Joint is a permanent Joint which is obtained by fusion of the edge of the two parts to be Jointed together, with Gry without the application of pressure and a filler material.

### Advantage:

- \* The welded structures are usually lighter tan
  than riveted structures.
- \* The welded Joints Provide maximum efficiency.
- + Alterations and additions can be easily made

- \* The welded provides very rigid Joints.
- \* The process of welding taxes Less time than

### Disadvardage

- \* It requires highly skilled Labour and supervision.
- \* The inspection of welding work is more difficult than riveting work.
- \* since there is an uneven heading and cooling during Fabrication.

# -> Welding Process.

- to welding Processes that use head along eg: Fusion welding Jayush
- \* welding Processes that use a combination of head and Pressure

Cg. Forge welding.

Re plastic defor

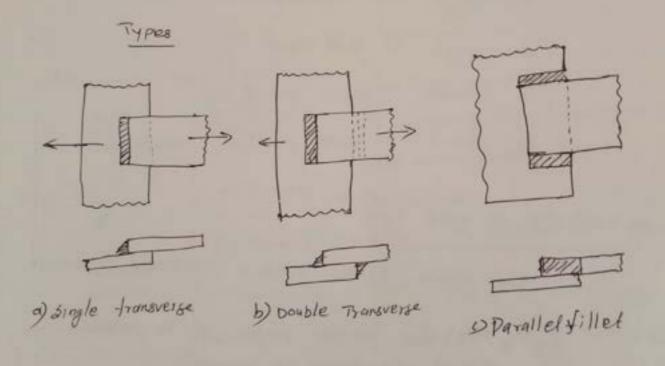
so lid stas

- 1. Lap Joint or filled Joint
- 2 Butt Joint

#### 1. Lap Joid

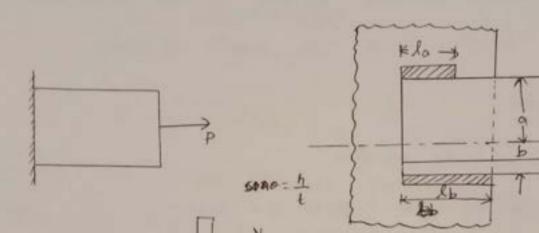
The lap Joint or the filled Joint is obtained by overlapping the plates and then welding the edges of the plate.

The cross - section of the filled is approximally triangular.



in axeral Load.

The butt weld Joint is subjected to tensile load 'P' the average tensile stress in the weld is given by



Shear Strees agens

T = Load

Area of weld

 $Z = \frac{P}{\sqrt{x+1}}$ 

Z= P COSHSXII

Z= P

he size of weld

la Length of weld in bottom

litotal weld cenyth: lat 1 5

t= hx sin45 P: Axial Load

t = hxwsus a : Distance of topweld foron grant

b: Distance of bottom weld from gravity Axis

Moment of topweld = lax fxa | M: Foreexors

Moment of bottom weld = lox+xla

Sum of moment of top & botom

laxxxa = lbxxxb

laxa = lbxb

W.k.t l-lailb

The butt count is ablanced by placing the

aguare bull Jard single v bult Jord

## stypes of bull Joint

1. square built Joint

2 single book V butt Joint

3. single U butt Joint

4. Double V butt Joint

& Double U butt Joint

## -> 5 election of word type

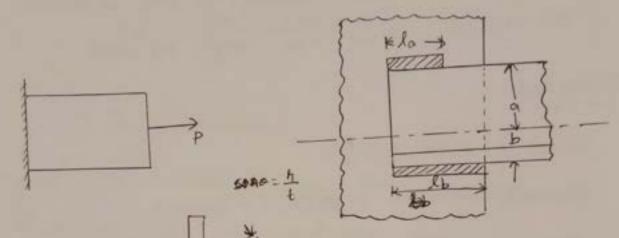
+ The shape of the welded component required

\* The thickness of the plates to be wolded.

\* The direction of the Forces applied.

to Design of Welded Joints based on strength of welding

The butt weld Joint is subjected to tensile load 'P' the average tensile stress in the weld is given by



Shear Slices sens

T = Load

Area of weld

h- size of weld.

ha Length of weldin Top

lb : length of weld in bottom

l=total weld sength = latlb

t= hx sin45 P: Axial Load

t = hxws45 a= Distance of topweld foron gravit

abils

b: Distance of bottom weld from gravity AxXX

Moment of topweld = lax fxa | H: YorkexOS

Moment of bottom weld = lox+xlo

dum of moment of top & botom

lasta = lostab
laxa = loxb

Wikit It loub

thick is to be welded by means of another place two parallel filled wold. The Platewas subjected to axial Load of 15 KN. The Maximum shear stress is not to exceed \$6 N/mm² - Find the length of weld.

### Given data:

width of place w=10mm

thickness of weld h = (61) + = 12.5mm

Monimum bhear stress Zmax = 56 N/mm²

Axial Load P= 50KN = 50X103 N

### To Find

Longth of Weld (1)

Avea.

Avea.

Tomax = P 1 = hx cos4s

Zmarx = P

56 = 56x10<sup>3</sup>

Dx 0-707 x 12.5 x L

LE SO BIRM

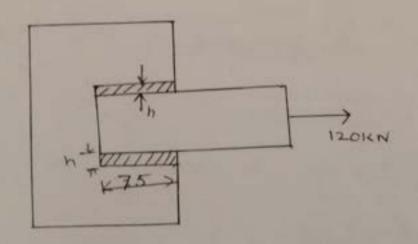
Method = < P54 0.B 113 >

0.707 P

56= 50x103 x 0.767

1 = 50.51 mm

2) Find the Size of weld for connecting as
shown in Fig. The Diensile Fore Force is acting
on Izokn. assume besign shear stress of
maderial 75 MPa.



Load P= 120 KN = 120 X18 N

Shear Stress Z = 75 MPa = 75 xxx N/mm2

length of weld L = 75 mm

To Find

size of weld h

Soln

Shear Stress = Load

Area of weld

Zmax = P = P

A sixl

76= 120×103

t = hx0645 =0.707h.

75 = 120x103 2 x 0.707h x75

h = 15.08 mm]

method -2 P64 11.3

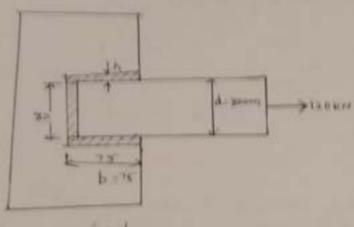
J = 0.707 P

75 = 0.707 x 120×103

Th=15.08 mm

as shown in tig. If the benefit went metring to land a permissible shows stress on the welding to 75 TSMPA

alven



wide w = Bomm

Thickness of place

Acral Local P = SOKN = SOXIBN 7207163

To Find

51 ze of weld h

Maximum Shear Gires: Local P

Areadweld 0-707h (261d)

1 1 1 0-707h

Es [h = 9.84 mm]

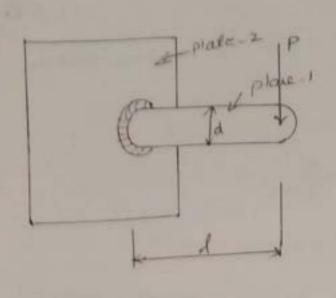
⇒ Type -2

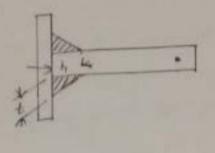
Eccentrically Loaded welded Joints.

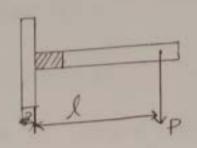
An eccentric Load may be imposed on welded to Joints in many ways. The stress induced on the Joint may be of different nature or of the same nature.

The induced stresses are combined depending upon the noduce of stresses.

when the Shear and bending street as are simultaneously present in a joint, then maximum streets are as follows







Step-7

Direct Stress Z= Load
Area of weld

27.17

5tep-2

Bending Stess 
$$\sigma_b = \frac{M_b}{I/y}$$
  $\frac{M_b}{I} = \frac{\sigma_b}{y}$ 

$$\sigma_b = \frac{M_b}{z} \left( \frac{27.17}{z} \right) \times \frac{1000 h}{z}$$

Mb: bounding of Moment = Loady Distance

Z = section modulus of weld &

Z = Zwxt 211.57

Zw: section modulus of weld shape

Zw in Psh D.B &11.6 >

Hep-3

Maximum Shear Stress

Zmax = 1 (06)2+472

P54.0B ×

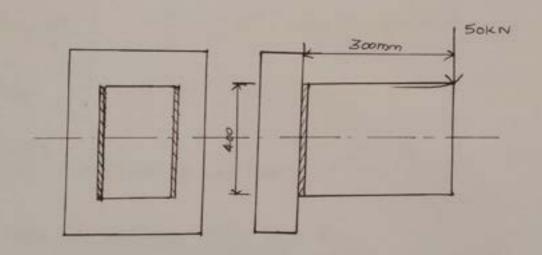
Formula For Area of weld in Diff cross section.

Area of Rectangular place = (2612d)+

Area of cylinder = Tidt

### Problems:

1. A bracked is welded to the vertical place by means of two filled welds as shown in Fig. Determine the size of weld the welds, if the Permissible Shear ofress is limited to 70 N/mm².



aven doda.

Load P=50KN

Permissible Shear Stress Zmax = 70N/mm

To Find 5128 of weld & h.

Step I

Area of weld = (2x 400) xt

+ 800 t t=0.707h

A= 800x0-707xh

Step.2 Bending Stress 06: Mb or 27.17

Mb bending Moment : Load x bistence

Mb = 50×103 × 300

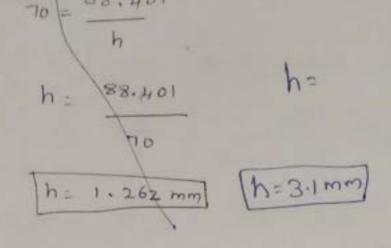
Z = Section Modulus

2 = Zwx + 2854.018 1115>

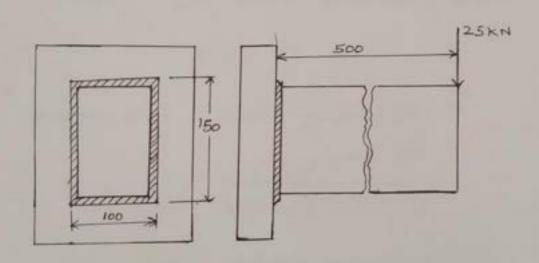
Step-3 Maximum Shear stress

$$70 : \frac{1}{2} \int \left(\frac{397 \cdot 1}{h}\right)^2 + 4 \left(\frac{82.4}{h}\right)^2$$

$$70 = \frac{1}{2} \sqrt{\frac{6.94^2}{897.8^2} + \frac{4 \times 88.4^2}{h^2}}$$



2. A beam of rectangular cross section is welded Support to a support by means of fillet welds as shown in Fig. Determine the size of welds, is the permissible shear stress in the weld 15 limited to 75 mm2



Given data

Load P = 25KN = 25x103N

depth d = 150mm width b = 100mm

Maximum Shear stress Zmax = 75 N/mm2 Length L= 500mm

To Find 6120 of weld.

A= 353.5 h

ME bending Moment : Load x Distance

Z= section Modulus

t= 0.707 h

$$\frac{D_{b}}{Z} = \frac{12.5 \times 10^{6}}{15.9 \times 10^{3} h}$$

$$\frac{D_{b}}{A} = \frac{12.5 \times 10^{6}}{15.9 \times 10^{3} h}$$

Step-3 Maximum shear stress

$$75 = \frac{1}{2} \times \sqrt{\frac{785.79}{h}^2 + 4 \times \frac{70.72}{h}^2}$$

$$75 = \frac{1}{2} \times \sqrt{\frac{785.79}{h^2} + 4 \times \frac{70.72}{h^2}}$$

$$75 = \frac{1}{2h} \times \sqrt{\frac{785.79}{h^2} + 4 \times \frac{70.72}{h^2}}$$

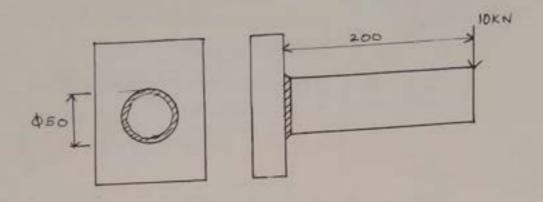
$$75 = \frac{1}{2h} \times \sqrt{\frac{785.79}{185.79}^2 + 4 \times \frac{70.72}{h^2}}$$

$$75 = \frac{1}{2h} \times 798.42$$

$$75 = \frac{399.21}{h}$$

$$h = \frac{399.21}{75}$$

3. A circular beam, 50 mm in diameter is welded to a support by means of a filled weld as shown in Fig. Determine the size of weld, if the Permissible shear stress in the weld is limited to 100 N/mm<sup>2</sup>



arven

bramater of shaft d = 50mmLength at shaft L = 200mmhoad  $P = 10kN = (0x10^3 N)$ 

Maximum Shear Stress Tmax = 100N/mm2

To Find

Size of weld (h)

soln

Step7 - M Direct Shear Stress

Z' = Load Area

Area: Trdt

A = 
$$47 \times 50 \times 0.707 \text{ h}$$

A =  $111.05 \text{ h}$  mm<sup>2</sup>
 $7 = \frac{10 \times 10^{3}}{111.05 \text{ h}}$ 
 $7 = \frac{10 \times 10^{3}}{111.05 \text{ h}}$ 
 $7 = \frac{90.04}{\text{h}}$ 

Bendung 64 vess  $7 = \frac{1}{10} \times \frac{1}{$ 

$$\frac{7}{5} = \frac{N_{h}}{2} = \frac{2 \times 10^{6}}{2776.38h}$$

$$\frac{7}{5} = \frac{7}{20.25}$$

$$\frac{7}{5} = \frac{1}{2} \times \frac{7}{5} + \frac{7}{4} \times \frac{7}{4}$$

$$\frac{100}{2} = \frac{1}{2} \times \frac{7}{20.25} + \frac{90.04}{h}$$

$$\frac{100}{2} = \frac{1}{2} \times \frac{7}{20.25} + \frac{90.04}{h^{2}}$$

$$\frac{100}{2} = \frac{1}{2} \times \frac{7}{20.25} + \frac{90.04}{h^{2}}$$

$$\frac{100}{2} = \frac{1}{2} \times \frac{7}{20.25} + \frac{90.04}{h^{2}}$$

$$\frac{100}{2} = \frac{1}{2} \times \frac{7}{100} \times \frac{7}{100}$$

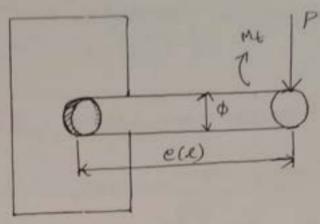
$$\frac{100}{4} = \frac{371.21}{100}$$

$$\frac{100}{4} = \frac{371.21}{100}$$

$$\frac{100}{4} = \frac{371.21}{100}$$

Type III

Design of weld by means of torque by end Load.



e = l = eccentricity or Length

P= Load

M1 = Twisting Moment

0 = diameter.

Procedure.

Step-I

Shear Direct Stress (or) Direct Shear Stress

ZI = P = Load

Area of Load

Step-TI

Secondary Shear Stress

 $Z_2 = \frac{M_{+} \times \gamma}{7} \quad \frac{P54}{27.17}$ 

Mt: Twisting Moment: Twisting Leaderd

7: Radius (d/2)

J - Jwx + 2 P54 11-5 &11-6>

Resultant Shear Stress

$$7 = \sqrt{(7)^2 + (72)^2}$$

Step-4 bending + stress 06

5 = Mbxy < PS9 7.1>

Jb= Mb = bending Moment

Mb = Load x bistance

Y= d/2 (07) half of depth.

Z = section Moderlus of weld

Z = Zwxt

Step-5

Maximum Shear Stress

(by) Permissible Stress

Johnax (Or) Zmax = 06 + 1 \ \ \tag{05} + 472

The bending stress N/mm2

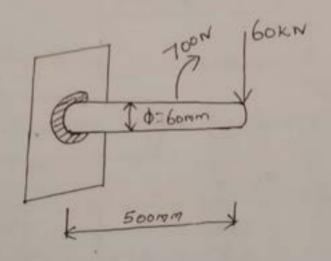
7 = Resultant Stress N/mm2

Problem

welded to a that place by a filled weld around the circumference of the shaft.

Roblem

1. The 60mm dometer of solid shaft one and is welded to support the the plate by means of filled weld. The other and end is Loaded at torque as shown in Fig. Find the size of weld. If the design shear stress of weld is 85mpg.



Given

diameter of shaft d= 60mm

Bending Load P= 60KN

Twisting Load PT = 700N

Length of solid shaft l= 500mm

5 50 h:

Step-1 - Direct Shear Stress

$$\frac{P}{Z_1} = \frac{L \cos d}{A r \cos of weld} = \frac{P}{A}$$

$$\frac{P}{T \cot t} = \frac{P}{T \cot t} = \frac{P}{T \times d \times 0.707 h}$$

$$V = \frac{d}{2} = \frac{60}{2}$$

J= Polar Moment of Inertia in weld

Step-3 Resultant Shear Stress

$$7 = \sqrt{\frac{450.226}{h}^2 + \left(\frac{5.25}{h}\right)^2}$$

$$7 = \frac{(450.226)^2}{h^2} + \frac{(5.25)^2}{h^2}$$

$$T = \sqrt{\frac{202.703 \times 10^3}{h^2} + \frac{827.56}{h^2}}$$

$$7 = \sqrt{\frac{(202.703 \times 10^{3}) + (27.56)^{2}}{h^{2}}}$$

$$O_b = \frac{M_b \times y}{T} = \frac{M_b}{Z}$$

$$Z_{\omega} = \frac{T \times 60^2}{4}$$

$$Z_{\omega} = 2.8 \times 10^3$$

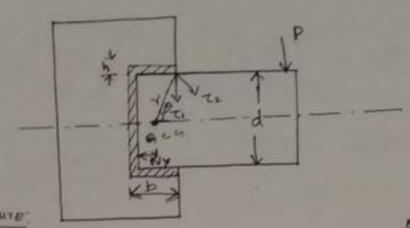
Step-5 Maximum Shear Stress

$$85 = \frac{15 \times 10^{3} / h}{2} + \frac{1}{2} \sqrt{\frac{15 \times 10^{3}}{h}^{2} + 4 \left(\frac{450.25}{h}\right)^{2}}$$

$$85 = \frac{7.5 \times 10^{3}}{h} + \frac{1}{2} \sqrt{\frac{225.81 \times 10^{6}}{h^{2}}}$$

$$85 = \frac{7.5 \times 10^3}{h} + \frac{1}{2} \times \frac{15.02 \times 10^3}{h}$$

$$35 = \frac{7.5 \times 10^3}{h^2} + \frac{7.51 \times 10^3}{h}$$



Procedure

Primary

Shear Stress Step-I bired shear stress or

ZI = Load

Area of weld

Step-II secondary shear stress due to twisting

Tz = Mtxx

Mt = Load x Eccentricity(er) Distance

r= Radius of Twisting r= Radius. (or) depth/2

J= Polar Moment of Inertia = Juxt

Ju value 1 n LP34 11.5 & +11.67

Step TII Resultant Shear Stress (or) Allowable (or)

Permissible shear stress

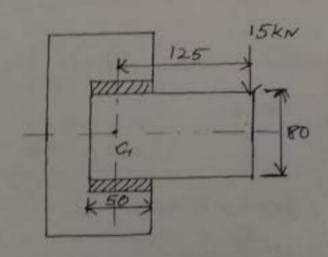
Zmax = \ 7,2 + 72 + 27,72 Cast

The AB= Radius con delpth 12

$$C_{18} = b - Ny$$
 $Ny \text{ Value} \quad \angle PSG 11.5 b 11657$ 
 $C_{05} = \frac{C_{18}}{7}$ 
 $O = \frac{G_{18}}{C_{05}} - \left(\frac{G_{18}}{7}\right)$ 
 $Y = \sqrt{AB^2 + G_{18}^2}$ 

### Problem:

I. A bracket comming a Load of 15km is to be welded as shown in Fig. Find the size of weld weld required if the allowable Shear stress is not to exceed 80 Mpg.



Chiven:

Load P= 15KN = 15x10 N

Allowable shear otress Zmax = 80MPa = 80N/mm2

depth of bar d = 80mm

To Find:

Size length of weld 'h'

Soln

Step ? Direct shear stress (or) Primary shear stres

Z1 = Load

Area of weld

 $Z_1 = \frac{P}{2b \times t} = \frac{15 \times 10^3}{2 \times 50 \times 0.707 \text{ h}}$ 

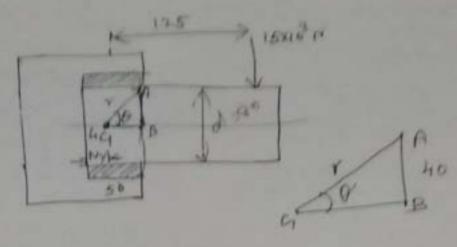
Z1= 212.16 h

Step-II secondary shear stress due to Twisting.

Mt = Load x Distance

Mt = 15x13x 125





not mention in the

so assume b/2

weld symbol

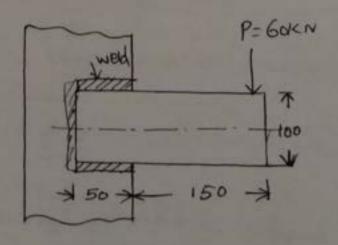
$$80 = \left(\frac{212.16}{h}\right)^{2} + \left(\frac{691.63}{h}\right)^{2} + 24\left(\frac{212.16}{h}\right)\left(\frac{691.63}{h}\right) \times 0.53$$

$$80 = \frac{45.01 \times 10^{3}}{h^{2}} + \frac{478.35 \times 10^{3}}{h^{2}} + \frac{155.54 \times 10^{3}}{h^{2}}$$

$$80 = 678.90 \times 10^3$$

$$80 = \frac{823.95}{h}$$
 $h = \frac{823.95}{80}$ 
 $h = 10.3 \text{ mm}$ 

2) A Rectangular steel plate 15 welded as a cantilever to a vertical column and supports a single concentrated Load P. as shown in Figure Determine the size of weld it shear stress in the same hot to exceed 140MPa.



Load P = 60KN = 60X103N

Width of with weld b = Looming

depth of weld d= womm

Maximum Bhear Stress Zmax = 140Mpg = 140N/mm2

To Find:

size of weld h'

boln

Step-I primary shear stress

ZI = Load

Area of weld

 $Z_1 = \frac{P}{(2b+d)t} = \frac{60 \times 10^2}{(2 \times 50 + 100) \cdot 0.707h}$ 

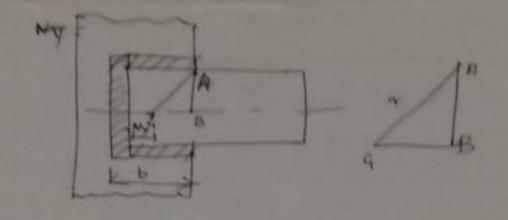
71= 424.32 h

Step-II secondary Shear Stress

 $Z_2 = \frac{M_t \times Y}{J}$ 

Mt = Load x eccentricity of from C.4

Mt = Load x [150+(b-Ny)] (or) Load x (150+40)



$$T = \int 56^{2} + 27.5^{2}$$

$$T = \int 56^{2} + 27.5^{2}$$

$$T = \int 56^{2} + 27.5^{2}$$

$$T = \int 30 \times t$$

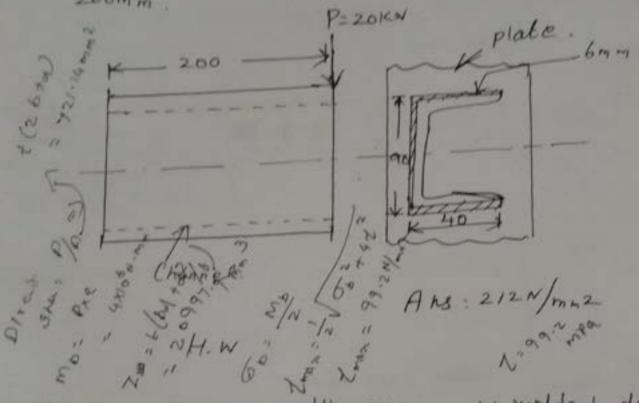
$$T$$

$$\cos\theta = \frac{GB}{\gamma} = \frac{37.5}{62.5}$$

$$\frac{140}{h} = \sqrt{\frac{424 \cdot 32}{h}^2 + \frac{2286 \cdot 32}{h}^2 + \frac{424 \cdot 32}{h} \frac{2286 \cdot 32}{h}^2} \times \frac{424 \cdot 32}{h} \times \frac{1226 \cdot 32}{h} \times \frac{1424 \cdot 32}$$

$$140 = \sqrt{\frac{2.49 \times 10^6}{h^2}}$$

3) Find the Maximum Shear stress du Indused in the weld of 6mm size when a channel as shown in Fig. 15 welded to a plate and Loaded with 20KN force at a distance of 200mm.

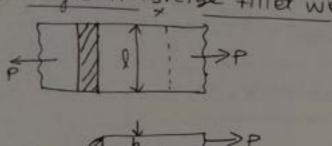


A plate of width 240mm 15 welded to a vertical plate by placing it on the vertical plate to from a Candilever with a projecting length of 480mm and overlap blu the plates as 120mm. Filled welding 15 done blu the plates on all the 7three side. A vertical Load of 35km 15applied on the cantilever at its 7rea end. parallel to the width. If the allowable stress on the weld is 94MN/m². determine the weld size

1415 - N

Strength of Transverse Fillet welded Joins

The transverse filled welds are designed for tensile strength, led us consider a single and double transverse filled welds as shown in Fig. > single transverse Filled welds



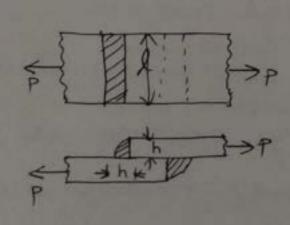
Maxi terallestress Othan = P

A= lxt

A= lx 0.707h

P= 0=x0.707 hxl

> Double transvense Filled weld:



Maxi Tensile Stress Of = Load

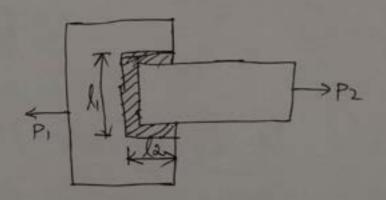
Max Area of weld

Otin = P A A= axlxt

P= 0+x A A= 2x 2x0,707 h

P= 5+2 2x0.707 hxl

Combination of single and double parallel weld



I, he length of single Filled weld

le = length of double Pralled Filled welt

Pi = Load on single transverse weld

Pz = Load on Double transverse weld

1 1 = d-t

Maximum Locad

d= death in mm t: thickness of weld (07) thick nees at plate

h=t

"Et Shear Strees

15 given P2 =

P= P1+P2

Pi= 0.707 hli xutnax

P2 = 2x 0.707 hxl2 x Ttmax (or) 2x0.707 xhxl2 x Tmax

P= 0.707 hlix Otnet 2x0,707 hlzx Otnes

Maxim pagemi

Permissible Stress of = 4099/ Of = Kt

Of = tensile stress

Ky = Stress Concentration Factor.

P= Load on tensile

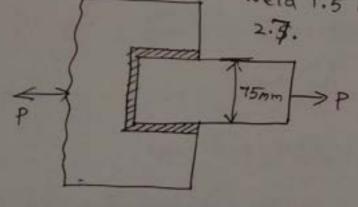
P= Ot Area of Plate

assume 15 Not given 5t=70 N/mm2

Tmax = 56 m/m2

Jointed with another plate by a single transverse weld and a double parallel filled weld as shown in Fig. The maximum tensile stress and shear stress are 70 Mpa and 56Mpa respectively.

Find the length of each parallel fillet weld,
if the joint is subjected to both source and
Fatigue Loading. Stress contration Factor for transverse weld 1.5 and parall Fillet weld 15
2.7.



Given:

depth of weld place d=75mm

thickness of weld place t=12.5mm

tensile stress of =70Mpa =70N/mn2

shear stress 7=56Mpa=56N/mn2

Stress Contentration factor kt forsingle fillet=1.5

Langth of single and Double Farallel weld

doln

1. Length of single Filled weld 2,

2. Length of double Parallel Fillet weld &2

If tobal Load P=P1+P2

$$\sigma_t = \frac{P}{A}$$

PI = Of Max X Area of single Fillelweb

If Kt is given Bt max = Ot Kt

h=t

A= 62.5 x0.707 x12.5

A = 552.34 mm2

PI= 46.66 x 552.34

P1 = 25 . 77 X10 N

Pz = Load on double parallel 7111ed weld

P2 = Streas x Area

P2 = 2× 0.707 × px l2 x Zmax

Tmax = T

Zmax = 56

hat

Zmax = 20.7 N/mm2

P2 = 2× 0.707×12.5×12×20.7

P2 = 365.87 l2

P= total Load

Stress  $\sigma_t = \frac{Load}{Area} = \frac{P}{A}$ of Plate

P= O+ X A

P= 70x dxt = 70x75 x12.5

P = 65.6 x 103 N

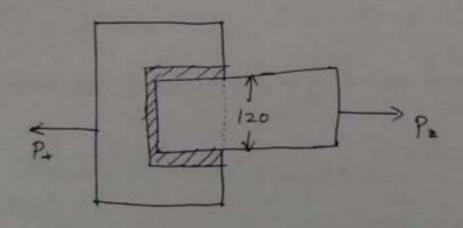
65.6x103 = 25.77x13 + 6 365.87 l2

$$365.87 l_2 = 65.6 \times 10^3 - 25.77 \times 10^3$$

$$l_2 = \frac{65.6 \times 10^3 - 25.77 \times 10^3}{365.87}$$

- 2) Determine the length of the weld run for a Plate of size 120mm wide and 15mm thick to be welded to another plate by means of
  - 1. A single transverse weld; and
  - 2. Double parallel weld Fillet welds when the Joint 15 Subjected to variable Load.

## Given:



depth of plate [wide] d= 120mm

thickness of plate t= 150mm

## To Find

Langth of single and Double transverge Filled weld.

Soln

1. Length of single Filled world &1

li= d-t

li= 120-15

R1= 105 mm

2. Langth of double transverse 7111et weld 12  $P = P_1 + P_2$ 

Pi Load on single transverse Fillet weld

PI = F A X TEMAX

Pi= lixt x Ttmax

OF- P/A

P= AxTtmax

Maximum tensile stress ofmax = 07

of is not given assume 70 N/mm2 h

Ke For single por Filled weld = 1.5

JE Max = 70

07 max = 46.66 N/mm2

Pi= 10.5 Pi= lix o. ToTxhx 5t max

h=+

PI= 105 x 0.707 x 15 x 46.66

P1 = 51.96x103 N

Pz = Load on double Filled weld

P2 = Areax Zmax

P2 = 2x l2 x 0.707 h x Zmax

Tmax = Maximum Bhear Stress

- Emax = E Kt To shear stress not given assume 56 m/mit and ky for double parallel filled weld = 27

Zmax = 56

Tmorx = 20.74 N/mm2

P2 = 2x = \$2x 0.707 x15 x 20.74

TP2 = t-099 x18 l2 N

P= Ot x Area of plate

P= 70x dxt = 70x 120x 15

P= 126x13N

P= PI+BZ

126 x18 = 51.96 x183 + 126 + 099 x18 /2

l2= 126 x103 -51.96x10

-1.099x18 439.89

l2=168.31mm

Welded to another place by means of Filled welds. The places are subjected to a Lead of soxus. Find the length of the weld so that the maximum stress not to exceed 56 Mpa. Ea consider the Joint First under static Loading and then under Fadigue Loading

H.W

4) A welded Joint as shown in Fig.

15 Subjected to an eccentric Load of ZKN.

Find the Size of weld, if the Maximum

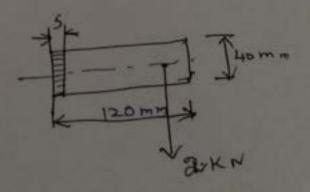
Shear stress in the Weld 15 25Mpa

\*\*

H.W

rote

A= 2xlx+



## DESIGN OF ENERGY STORAGE IN DEVICES

Spring!

Spring is the elastic body whose bunction is to loaded on spring compression and to recover the osciginal Shape when load is removed.

Application of spring:

\* Absorb or control energy due to either shock on vibration as in the care spring.

\* Railway Buffers.

\* Airwraft

Types of sporing:-

\* Helical spring .

Helical spring.

\* compression coil helical spring

\* Terrsion coil helical sporting.

condeat sprias

\*Conical spring

at Tooksional spacing

Heat spring or laminated spring

\* Dioc Spring

\* special purpose spring

DESIGN PROCEDURE OF HELICAL SPRING !-

Step 1: Diamension of spring. Refor DE 19:7.100)

d-Dianeter to wire

D-Mean diameter of Spring

Do-Outer diameter of spring

Do = D+d

Shear storess  $T = \frac{8PD}{\pi d^3} = \frac{8PC}{\pi d^2}$ 

P-axial load.

It two load is given, P= Proax

Ks-Wahl stress factor

 $k_s = \frac{4C-1}{4C-4} + \frac{0.615}{C}$ 

c - sporing index

 $C = \frac{D}{d}$ 

Ostop z : Number of specionag active coils (n)

By using the deflection relation.

 $Y = \frac{8PD^3n}{Gd^4} = \frac{8PC^3n}{Gd}$ 

where

n-number of coids

Gr-madulus of rigidity.

Gr = 8 X104 N/mm

P = load

To two load is given P = Pmax -Pmin.

otep 3: Stiffness of spring 'q'.

 $Q = \frac{Gd^4}{8D^3n} = \frac{Gd}{8C^3n} \quad (or) \frac{y}{w}$ 

The A: Total number of coil (nt)

From Data book pg.no.7.101

For and condition, plane or plane ground

Step 5: Solid Length of Spring

Le = dn +d

Step 6: Face length of spacing

- Ly = Pn + d. (or) Pitch value.

Ly = Ls+Y

Step 7: Pitch of coils.

 $P = \frac{L_f - L_s}{n_t} + d$ 

Step 8: Helix angle 'K'.

 $x = \tan^{-1}\left(\frac{P}{\pi D}\right)$ 

P-Pitch value

Step q'workdone on energy. 27.100>

$$U = \frac{PY}{2}$$

Note: -

Workdone (091) energy  $\times$  No of Spring =  $\pm E$   $\frac{PY}{2} \times n = \frac{1}{2} mv^2$ 

1. Derive the expression for Shear Stress

for helical sporing by using torrsional

where 
$$J = \frac{\pi d^4}{32}$$

$$R = \frac{d}{2}$$

$$\frac{P \times (D/2)}{\binom{n \cdot d^4}{32}} = \frac{T}{(d/2)}$$

$$T = \frac{8PD}{\pi d^3}$$

$$T = k_S \frac{8PD}{Rd^3}$$

A helical coil sporing is to be designed for a operating load of range is 90N to 135 N. The deflection of the Spring for the load range is 7.5 mm. Assume the Spring Index of 10. The pormissible Shear Stross of 480 N lmm² and modulus of rigidity is 0.8\$ x10 N/mi For the material design the spring.

Given Data!

Pmin = 90 N Pmax =135 N

y = 7.5mm

C = 10

I = 480 N/mm2.

G = 0.8 × 105 N/mm2

To Find! -

Design of Helical Spring.

Solution! -

Refer DB Pg. 7.100

Step 1: Dimension of Spring.

d-diameter of wisce

D-Mean diameter of Spring

Do-Outon diameter of Spring

 $D_0 = D + d$ 

T = Kg 8PC

$$P = P_{\text{max}} = 135 \text{ N}$$

$$K_{\text{S}} = \frac{4C - 1}{4C - 4} + \frac{0.615}{C}$$

$$= \frac{4(10) - 1}{4(10) - 4} + \frac{0.615}{10}$$

$$K_{\text{S}} = 1.14$$

$$T = 1.14 8 \times 135 \times 10$$

$$480 = 3919.03$$

$$d^2$$

$$c = \frac{D}{d}$$

$$D_0 = 33 \text{ mm}$$

Step 2: Number of active coils (n)

By using the deflection relation

$$y = \frac{8PD^3n}{Gd^4}$$

P= 
$$\frac{P}{max}$$
 -  $\frac{P}{P}min$  = 135-90  
= 45 N.  
N =  $\frac{7.5 \times 0.8 \times 10^{5} \times 3^{4}}{8 \times 45 \times 30^{3}}$   
P =  $\frac{610^{4}}{80^{5}n}$   
=  $\frac{0.8 \times 10^{5} \times 3^{4}}{8 \times 30^{3} \times 5}$   
Step 4: Total numbers of coil ( $n_{t}$ ).  
Por and condition,  
N<sub>t</sub> =  $\frac{600^{4}}{80^{5}n}$   
Assume square and Ground Condition.  
N<sub>t</sub> =  $\frac{1}{12}$   
= 5+2  
 $\frac{1}{12}$   
Step 5: Solid Length of spring.  
Consider square condition.  
 $\frac{1}{12}$  =  $\frac{1}{12}$   
=  $\frac{1}{12}$   
= 3(5) + 3(3)  
 $\frac{1}{12}$ 

Promise Commence (FD Com

Step 6: Free length of spring.

$$L_{5} = Pn + \frac{1}{2}d$$
 $L_{5} = Pn + \frac{1}{2}d$ 
 $L_{5} = Pn + \frac{1}{2}d$ 
 $L_{7} = L_{5} + V$ 
 $= 24 + 7.5$ 

[14] = 31.5 mm

Otap 7: Pitch of coil

 $P = \frac{L_{7} - L_{5}}{h_{4}} + d$ 
 $= \frac{24 + 31.5 - 24}{h_{4}} + \frac{3}{4}$ 
 $= \frac{4.07}{\pi \times 30}$ 
 $= \frac{4.07}{\pi \times 30}$ 

Step 9: workdone or energy -  $U = \frac{PV}{2}$ 
 $= \frac{13.5}{2} \times 7.5$ 
 $= \frac{13.5}{2} \times 7.5$ 

U=506.25 N.mm

2. A Rail wagen moving at a velocity of 1.5 m/s is brought to rest by a bumber consisting of two helical springs arranged in parallel. The mass of the wagon is 1500 kg. The spring are compressed by 150 mm in bringing the wagon to rest. The spring index can be taken as 6. The spouring are made of oil hondered and tempered steel wire with ultimate tensile Strength of 1250 N/mm² and modulus of origidity of 81,370 N/mm2. The posinissible Shear Stress for the spring wire can be taken as 50%. of the ultimate tensile strength. Design the spraing and calculate 1) where diameter ii) Mean coil dlameter iii) Number of active coils. iv) Total number of coil V) Solid length vi) Free length

vii) pitch of coil.
viii) Required spring rate and
ix) Actual spring rate.

Given bata: -

V =1.5 m/s

m = 1500 Hg

no of spring, n = 2 C = 6

V = 150 mm. Ou = 1250 N/mm²

G = 81370 N/mm2

7 = 50 % Ou

T=0.5 Ou =0.5 x1250

=625 N/mm2

To Find! -

Design of Holical Spring:

solution! -

Step: Dimension of Spring

d-diameter of wione

D-Mean wil diameter of spring

Do -out

Solution:

Stop 1 ! Dimension of Spring.

d-diameter of wire

D-Mean diameter of Spring.

Do-outer diameter of Spring.

$$T = \frac{1}{8} \frac{8PC}{\pi d^2}$$

$$k_s = \frac{4c-1}{4c-4} + \frac{0.615}{c}$$

$$= \frac{4(6)-1}{4(6)-4} + \frac{0.615}{C}$$

$$\frac{PY}{2} \times n = \frac{P \times 150}{2} \times 2$$

$$d^{2} = \frac{1.25 \times 8 \times 11-25 \times 10^{3} \times 6}{71 \times 625}$$

$$C = \frac{D}{a}$$

$$D_0 = D + d = 120 + 20$$

Step 2: Number of active cails 'n'

$$y = \frac{8PD^3n}{Gd^4}$$

$$D = \frac{150 \times 81370 \times 20^4}{8 \times 11.25 \times 10^3 \times 120^3}$$

Step 3: Stiffness of spring!

$$9 = \frac{\text{Grd}^4}{8 \text{Din}}$$

$$= \frac{81370 \times 20^4}{8 \times 120^3 \times 13}$$

stop 4: Total number of coils.

Rober DS 1701.

Assume square condition

step 5: Solid Length of Spring

Step 6: Free length of spring.

Step 7: pitch of coils.

$$P = \frac{L_f - L_s}{n_t} + d$$
 =  $\frac{L_{70} - 320}{15} + 20$ 

Required spring reate: -
$$q_R = \frac{p}{y}$$

$$= \frac{11.25 \times 10^3}{150}$$

$$q_p = 75 \text{ N/mm}$$

3. A helical composession sporing is used to absorb the sa shock. The initial compression of the spring is somm and it is buither compressed by somm while absorbing the shock. The sporting is to be absorb 250 Joule of energy during the pricess. The spring index can be taken as 6. The sporing is made of pattorned and cold drawn steel wire with an utimate tensile strength of 1500 MPa and Modulus of suigidity as 81370MPa. The permissible shear stress for a spouring wire should be taken as soy Ultimate tensile strength. Design the spring and calculate juice diameter ii) Mean coil diameter

iii) Numbon of Active coils iv) Free length v) Pitch of the turn

Given Data! -

(IJ=INIm)

y som

initial compression, y, = 30 mm.

Final Comprussion, 42 = 80 mm.

Energy absorbed, U = 250 J

= 250 × 103 N mm

C = 6 .

ou = 1500 N/mm2

G1 = 81370 MPa

=81370 N/mm2

T = 30% Gu

=0.3 × 1500

=450 N/mm2

To Find ! -

iswire diameter

ii) Mean coil diameter

iii) Number of Active coils

iv) Force length

V) Pitch of the turn

Design of Helical spring.

Solution!

Otop 1 : Dimensions of Spring:

d- Dimension of spring

D.- Mean Diameter of Spring

Do-outer diameter of spring

$$T = Ks \frac{8PC}{Kd^2}$$

$$Ks = \frac{4C-1}{4C-4} + \frac{0.615}{C}$$

$$= \frac{4(6)-1}{4(6)-4} + \frac{0.615}{6}$$

15=1.2万

Find P.

To two deflection and energy is

given y = 30 mm

stiffness  $q_i = \frac{Load}{deflection} = \frac{P_i}{y_i}$ 

 $Q = \frac{P_L}{30}$ 

P, =30×9

y2 = 80 mm.

 $Q = \frac{P_2}{80}$ 

P2 = 80×9

Average Force during compression = PAP

= 309 + 869

= 1109

= 559

Energy absorbed during Shock & Allere

= Average X Second comprossion
Force of Spring.

250 ×10 = 9559 × 800 50

$$P_1 = 30 \times 90.9$$

P2 = 80 x 90.9

P = Pmax = 7272 N.

d +18.75 mm

say d=20 mm

 $C = \frac{D}{d}$ 

D =100 mm

Po = D+d =100 +20

The Ram

=140 mm.

obtep 4: Number of active coils:

$$y = \frac{8PD^3n}{Grd^4}$$

$$P = Prox - Pmin$$

$$= 7272 - 2727$$

$$= 4545 N$$

$$n = \frac{30 \times 31570 \times 20}{8 \times 4545 \times 86^3 \text{ s}}$$

$$n = 16.57$$

$$9 = Grd$$

$$9 = Grd$$

$$9 = Grd$$

$$8c^3n$$

$$9 = \frac{81370 \times 20}{8 \times 6^3 \times 17}$$

$$9 = 55.39 \text{ N/mm}$$

$$9 = 55.39 \text{ N/mm}$$

$$9 = 55.39 \text{ N/mm}$$

$$9 = 57.39 \text{ N/mm}$$

$$9 = 57.39 \text{ N/mm}$$

$$9 = 57.39 \text{ N/mm}$$

Step 5: Bolid length of Spring
$$L_S = dn + 3d = (20 \times 19) + (3 \times 20)$$

$$L_S = 440 \text{ mm}$$

Step 7: Pitch of coils

$$P = \frac{L_4 - L_5}{n_E} + d$$
 $= \frac{520 - 440}{19} + 20$ 
 $P = 24 - 4mm$ 

Stop 8: Helix angle.

$$\alpha = \tan^{-1}\left(\frac{P}{\pi D}\right)$$
 $p - pitch value$ 

$$= \tan^{-1}\left(\frac{2u \cdot 4}{3 \cdot 14 \times 120}\right)$$
 $\alpha = 3.70$ 

H. Dessign a helical spring for a spring land sofety valve for the following condition

Diameter of valve Seat = 65 mm.

Operating pressure = 0.7 N/mm²

maximum pressure when the valve blows of breely = 0.75 N/mm²

Maximum lift of the valve when the pressure graises from 0.7 to 0.75 N/mm² =

Maximum allowable stress = 550 N/mm² Modellus of rigidity = 84 KN/mm²

opping index =6.

Draw a neat stretch of a free spring showing the main dimensions.

Gun Data:-

Diameter of valve stat Di=65 mm

Pmax =0.75N/mm²

Rmin = 0.7 N/mm2

T = 550 N/mm2

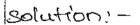
G1 = 84 KN/mm2

=84 X103 N/mm2

C = 6. g = 3.5 mm

TO Find! -

Design of helical spring



Stop 1: Dimension of spacing (Rejor DB pg -7-100

d-diameter of wire

D - Mean diameter of Spring

Do - outer diameter of spring

T=Ks8PC 97d2

$$\frac{1}{4c-4} = \frac{4c-1}{4c-4} = \frac{0.615}{6}$$

$$= \frac{4(6)-1}{4(6)-4} = \frac{0.615}{6}$$

$$d^2 = \frac{1.25 \times 8 \times 2488 \times 6}{77 \times 550}$$

$$d = 9.29$$

$$d = 10 mm$$

$$C = \frac{D}{d}$$

$$D_0 = D.+d$$
 $= 60 + 10$ 

$$y = \frac{8PD^3n}{Gd^4}$$

$$n = \frac{3.5 \times 84 \times 10^{3} \times 10^{4}}{8 \times 165.93 \times 66^{3}}$$

$$9 = \frac{Gid}{8c^3n}$$

$$=\frac{8 \times 6^{3} \times 10}{8 \times 10^{3} \times 10}$$

Step 4: Total no of coils he'

Assume square condition

Stop 5: Solid length of Spring

$$= (10 \times 11) + (3 \times 10)$$

Step 6: Free length of 8 pring

Step 7: Pitch of coils

$$P = \frac{L_f - L_S}{n_E} + d$$

$$= \frac{143.5 - 140}{13} + 10$$

Step 8 ! Halix angle

$$x = \tan^{-1}\left(\frac{p}{\pi D}\right)$$

$$= \tan^{-1}\left(\frac{10.26}{\pi \times 60}\right)$$

Step 9: workdore or energy

$$U = \frac{P_{\text{max}} y}{2}$$
= 2488.73 × 3.5

DESIGN OF SPRING SUBJECTED TO UNDER

Its Factor of safety is given, Find the coil diameter d

( Refer DB Pg 7.102)

 $\frac{1}{n} = \frac{T_m - T_a}{T_y} + \frac{2T_a}{T_e}$ 

where Im - mean shear stress

 $\frac{T_{m} = 8 k_{sh} P_{mD}}{\pi d^{3}} = \frac{8 k_{sh} P_{mC}}{\pi d^{2}}$ 

Ta - amplitude shear stress

 $T_{a} = \frac{8k_{s}P_{a}D}{\pi d^{3}} = \frac{8k_{s}P_{a}C}{\pi d^{2}}$ 

Pm = mean load

= Prox + Pring a

Pa = amplitude load

= Pmax - Pmin

ks = wahl steens factor.

 $= k_{Sh} k_{C}$  (or)  $k_{S} = \frac{4C-1}{4C-4} + \frac{0.615}{C}$ 

Ksh = direct shear bactor

 $k_{Sh} = 1 + 0.5$ 

Kc = curvature Factore.

To = andurance shear stress

=ansite

20.5 Ty

where ty = Yield shear stress

Ty = 0.405 Ou (For light service).

Ou = ultimate tensile strength.

A helical compression spring made of oil tempered carbon steel is subjected to a load which varies from 400 N to 1000 N. The Spring index is 6 and pesign factor of safety is 1.25. To the yield stress in shear is 770 MPac and endurance rotress in Shear is 350 MN/m². Find the Size of Spring wire, diameter of the spring, Number of turns of the spring, Number of turns of the spring and free lingth of Spring.

The compression of the Spring at the maximum load 30mm. The modulus of regidity for spring material may be taken as 80 KN/min and Design the helical spring.

Gin Data: -

Pmax = 1000 N

Pmin = 400N

Fos (or) n =1.25

C = 6: , Kc = 1.15

Ty = 770 MPa = 770 N/mm²

Te (er) To = 350 MN/m² = 350 X10 N/m² = 350 N/mm²

y = 30mm

G = 80 KN/mm2

=80×103 N/mm2

To Aind! -

i) Size of spring wire.

ii) diameter of spring

iii) Number of turns of spring

14) Free length of spring

Solution! -

estep 1 : Dimension of sporing.

If factor of safety is given

 $\frac{1}{h} = \frac{T_m - T_a}{T_y} + \frac{2 T_a}{T_e}$ 

(Refer DB Pg No. 7.102)

$$T_{m} = \frac{8 \text{Ksh Pm C}}{\pi d^2}$$

$$K_{\rm Sh} = 1.08$$

$$T_{m} = \frac{8 \times 1.08 \times 700 \times 6}{\pi \times d^{2}}$$

$$T_{\rm m} = \frac{11550.82}{d^2}$$

$$T_a = \frac{8 \text{ KsPaC}}{7 \text{ d}^2}$$

$$Pa = \frac{P_{\text{max}} - P_{\text{min}}}{2} = \frac{1000 - 400}{2}$$

$$T_{\alpha} = 8 \times 1.24 \times 300 \times 6$$

$$\pi \times d^{2}$$

$$T_a = \frac{5683.74}{d^2}$$

$$\frac{1}{1.25} = \frac{11550.82 - 5683.74}{d^2 \times 770} + \frac{2(5683.74)}{d^2 \times 350}$$

$$\frac{1}{1.25} = \frac{40.09}{d^2}$$

$$d^2 = 40.09 \times 1.25$$

$$d = 7mm$$

$$C = \frac{D}{d}$$

$$D = 6 \times 7$$

Step 2: Number of active coils, n

$$y = \frac{8PD^3n}{Grd^4}$$

$$n = \frac{30 \times 80 \times 10^3 \times 7^4}{8 \times 600 \times 42^3 \text{ k}}$$

Step 3: Stiffness of spring

$$q = \frac{G_1 d}{8c^3n}$$

$$=\frac{80 \times 10^3 \times 7}{8 \times 6^3 \times 16}$$

9=20.25 N/mm

Step 4: Total number of coils, of ne

Assume square condition

Le = dn + 3d

$$=(7\times16)+(3\times7)$$

Ls = 133 mm

$$= 133 + 30$$
 $4 = 163 mm$ 

$$P = \frac{14 - 15}{n_t} + d$$

$$= \frac{163 - 133}{18} + 7$$

$$\alpha = \tan^{-1}\left(\frac{P}{ND}\right)$$

$$= \tan^{-1}\left(\frac{8:66}{3.14 \times 42}\right)$$

$$\alpha = 3.75^{\circ}$$

$$U = \frac{PY}{2}$$

$$U = \frac{1000 \times 30}{2}$$

Multi leab spring are used to suspension of bostuque, toucks and nailway Multi leab spring consist of a series of blat plates. The Flat plates are called leaves of the Spring.

The leaves are graduated length.

The leaf at top has maximum length. The length gradually decrease from the top leaf to bottom leaf.

Lead acting on the spring = 2P

Ratio of Total depth and breadth = 18t nt b

where, n - number of leaf

t-thickness of spring

b-breadth of leab

Formula used (Reber DB Pg. 7.104)

Bending stress  $\sigma_b = \frac{6PL}{nbt^2}$ 

Deblection of the spain  $y = \frac{6PL^3}{Enbt^3}$ 

the material.

initial space,  $h600x = \frac{2PL^3}{Enbt^3}$ Nip(h)

Load excerted on the clipping

Bolt Assembled,  $P_b = \frac{2n_e n_g P}{n(2n_g + 3n_e)}$ 

un = ne + ng where, ne - number of extra bull length leaves

ng - number of graduated leaves

1. A locomotive semi abinaisated laminated sporting has an overall length of I'm and substairs a load of TOKN at its centre. The spring has three full length leaves and 15 graduated leaves with a central band of 100 mm width. All the leaves are to be stressed to 400MPa. when full loaded reation of total spring depth to that of width is 2 - The young's modulus of 210 KN/mm Determine i) thickness and width of leaves ii) The initial gap that Should provided between the full length and graduated leaf band load is applied

(iii) The load executed on the band after the spring is assembled.

$$\frac{nt}{h} = 2$$

To Find! -

- i) Thickness and width of leaves.
- ii) Initial gap provided between full length and graduated leab (x)

Solution! -

i) Thickness and width of leaves

$$\frac{nb}{b} = 2$$

$$\frac{18t}{b} = 2$$

$$b = 9t$$

If Bending Load is given

$$\sigma_b = \frac{6PL}{nbt^2}$$

$$400 = 6 \times 35 \times 10^3 \times L$$

21 = 1000 - Length of central band

= 1000 - L

= (000 - 100

2L =900

L=450 mm

 $t^3 = \frac{6 \times 35 \times 10^3 \times 450}{18 \times 9 \times 400}$ 

E =11.34 mm

say t=12mm

₽ b = 9t = 9x12

b=108 mm

$$x = \frac{2PL^{3}}{Enbt^{3}}$$

$$= 2 \times 35 \times 10^{3} \times 450^{3}$$

$$= 210 \times 10^{3} \times 18 \times 108 \times 12^{3}$$

iii) boad exerted

$$P_{b} = \frac{2 n_{e} n_{b} P}{n(2 n_{g} + 2 n_{e})}$$

$$= \frac{2 \times 3 \times 15 \times 35 \times 10^{3}}{(8(2 \times 15) + (3 \times 3))}$$

$$P_{b} = 4487.17 \text{ N}$$

2. Design a leaf spring for a truck to the following specification. Maximum lead of spring is 140 kN, Number of Spring = 4, Material = Chromium, vanadium, spring pounissible tensile stress boompa, Maximum number of leaves = 10, span of spring=1000mm; permissible deflection = 80 mm, Young's modulus = 2 × 10<sup>5</sup> MPa.

#### Solution! -

i) Thickness and width of leaves.

$$\frac{nt}{b} = 2$$

$$\frac{b}{b} = 0.2$$

$$O_b = \frac{6PL}{rbt^2}$$

DE LECO

600 = 6×70×10 +500

+3 = 6×70×103×560

t = 18.60 mm

h= 93.00 mm/

i)  $\chi = \frac{9PL^3}{ED6E^3} = \frac{8 \times 70 \times 10^2 \times 1160^3}{2 \times 10^5 \times 10 \times 93.02 \times 18.8^2}$ 

 $P_b = \frac{2 \operatorname{re} \operatorname{ng} P}{\Gamma(2 \operatorname{rg} + 3 \operatorname{re})}$ 

\$\$

### UNIT-5

# DESIGN OF BEARINGS.

## Blearing! -

Bearing is a mechanical elements
that permit relative motion between two
Ports such as the shaft and housing with
minimum friction.

#### Functions! -

\* Force orotation of shaft

holds it in the correct position.

\* The bearing takes of the forces that act on the shaft.

## chassification ! -

Depending upon the direction of the load \* Radial bearing

-Load act on the beauting

Perpendicular to the direction of motion

of the moving element.

\* Thoust Bearing.

- The land act along the and

of retation.

ii) Depending upon the nature of contact \* sliding contact bearing

The slides take place the swifaces of contact between the moving dement and the fixed element.

\* Rolling contact bearing

interpost between the moving and fixed element.

Types of sliding contact bearing \* Full Journal bearing (360)

\* Partial Jeweral bearing (120° or 180)

Hydro dynamic Lubricated Bearing:

The thick flim becomings are those, in which the working surface are completely separated from each other by the lubricant.

\* Bounday Intricated Bearing

- Thin film bearings partial
contact.

\* . . zero film Bearing - without any lubricant.

DESIGN PROCEDURE OF JOURNAL BEARING!

Step 1: Dimensions of bearing (D, L).

(Folgo DB pg. 7.31)

L = ?

\* The machine is not given assume centrifugal pump

\* Diameter of Javard is not given Assume D = 100 mm.

x. Power is given  $p = \frac{2\pi NT}{60 \times 1000}$ 

\* Torque  $T = \frac{\pi}{16} \cdot D^{3}$ ; D = 9

Step 2: Pressure developed (or) Check 6007 Bearing pressure.

\* Allowable Bearing pressure  $[P_{a}]_{all}$  in  $kgf/cm^2$ 

Bearing, Ind A DXL

Andu < [P]all

Design is safe.

Step 3: Viscosity of Oil. (=) (Peyes, DB og 7.81)  $\frac{Z_n}{D} = ?$ where, Z in Cps (centi poise) P = Pinducad n-speed in spm. Step 4' Belect grade of oil (SAE) (Reper DBP NO 7.41) \* to = operating temperature in bearing \* To to is not given assume bockoga: Step 5: Bearing module: -From basic theory  $\frac{Z_n}{P}$   $\frac{1}{2}$   $\frac{Z_n}{P}$ . Bearing operated on hydro lubrication. step 6: coefficient of frictio M. (Refair DBI pg. No. 7.34) MckEES Equation.  $= \frac{33.25}{10^{10}} \left( \frac{20}{p} \right) \left( \frac{D}{c} \right) + K$  $\left[\frac{Zn}{P}\right]_{11}$   $\Rightarrow$  (Refer DB Pg. 7.31)

D-Diameter of Journal in com C-Diameteral Cleavance (Refer DB M 7.32 C (diameteral clearance) in micron from table pg. No. 27.32

K is constant from the graph.

(Pg. NO. \*\* 7.34).

Step 7: Heat generated. (Hg) (08) Power Lost

(Refer DB Pg. 7.34).

Heat generated Hg= HWV in Kgfm/min.

N- had in kgf V- velocity in m/min.

V=TDN m/min.

Step 8: Heat dissipated, Hd (Res DB Pg. 7.34

Ha = (At +18) LD kg+ m (min

where, L and D in meter

K-Constant heat dissipation (Pg.7.35).

Dt - average temperature suse

 $\Delta t = \frac{t_0 - t_a}{2}$ 

 $H_d = CA(t_o - t_a)$ 

```
Step 9: Artificial Cooling is required or not?
        Ib the > the , Axitifical cooling is
required
Step 10: mass of oil (m)
          m = Hg-Hd in kg-f/min
    where, c'-specific heat of oil
                                  (DB Pg . 7.34)
             -17100 kgf cm/kgf°c
            = 171 kgfm/kgf°c
        Oto - increase in tempocature of oil
   Ih it is not given sto = 10°C
Step 11: selection of Bewing material
                     Refer DB Pg.7.30
Step 12: Diameter of bearing Do
      Db = D+C
where, D - Diameter of Journal
        c - Diameteral clearance
Note: -
     * INS/m2 = 1 kg/ms (or) Pas
    * ICP = 1 XIO Pas
```

= 10 9 MPas.

1

ප

05

W



# NOTES

NOTES
Marin curve! - Gurbon curve is a parabola drown
between endurance limit and ultimate tensile strongth.
Sodorberg line: Line Joining by on mean stross axis
and on Stross amplitude axis is colled sortered line strain used for dutile material steadman line: I are Johning on on mean stross axis and
on Stross amplitude axis is called as Gwardman line
alagerum used for brittle material
Method to roduce stress concentration sort partitions  * * Avoiding Shorp corners
* Avoiding Sharp corners Barker South
* 4 uso of multiple holes instead of Single holes
Xundorcutting the Shoulder parts.
Teline .
Shaft runs so that the additional deflection of the
Shaft from the axis of motation becomes infinite
is known as critical or whirting spead
Plinitation of welding!
*It has poor vibration damping characteristics
+ wolding results distortion of pasts which
induces residual strosses
afine and born with a heard integral to it. The
cylinder portion of the nivet is called shark or body and
lower portion of the Shark is known as tail. The
riveted Joints are widely used for Jointing light metal

## NOTES

6) Self locking of power some Probatives It the priction angle (b) is greater than helix angle (a) of the power snew, the torque required to las the load will be positive which indicates an apport applied to large the load. This type of screw is troops Self-laking screw. The efficiency is less tran 50%. 7) Pitch : It is the axial distance from a point on one thread to corresponding point on the next throad Least !- It is the distance moved by the scrow in a one from (8) By what material threaded fasterious are made? Steel is the mitorial of which most of basteness are made For improving their proporties alloy Stool like nicked glood, Ni-Cr Stool, Cr-V Stool are a) caulking: The edge of one plate is prosed tightly against the plate on which it rests by means of a caulking tool Fullering! - The operation is similar to coulting operation except that fullering makes used of a tool baving thickness at flue end equal to the plate thickness and it carries an angle of 80 at the end 6) DB coupling & dutch: A coupling is a device us to make pormanent or semi permanent connertion a clutch permits rapid connection or disconnection at will of the operator.

1. Design a Journal bearing for a centrifugal pump for a following Data Diameter of Journal = 150 mm Load of Bearing = 40 KN Speed of Joweral = 900 pm. Gin Data: D=150 mm = 0.15 m = 15 cm W=40KN =40X(03N) =40 x102 kgf N = 900 xpm Solution: -Step 1: Dimension of bearing For centrifugal pump (Refer DB pg. 7:31)  $\frac{1}{D} = 1.0 - 2.0$ Take = = 1.5 L=1.5X15

 $L = 1.5 \times 15$  L = 22.5 cm

OSTEP 2: pressure developed on check for bearing pressure.  $[P]_{all} = 7-14 \text{ kgf/cm}^2 \qquad \text{(Refer Pg. 7.31)}$   $Take [P]_{all} = 14 \text{ kgf/cm}^2$ 

 $P_{ind} = \frac{W}{D \times L} = \frac{40 \times 10^2}{22.5 \times 15}$ 

=11.85 hgf/cm2

Pinduid Z[P]all , Design is safe

$$\frac{Z_0}{P} = 2844.5$$

$$\frac{2}{2} = \frac{28141.5 \times 11.85}{900}$$

Step 4: Delection of grade (Refer DB 19.7,41)

From the graph, x-axis temperature

assume 60°c to 90°c

Take to = 60°C

Y axis Z = 37.45 Cps SAE 20

Grade of oil SAE 40

is selected.

Step 5: Bearing module.

$$\frac{37.45\times900}{11.85} > \frac{1}{3} (2844.5)$$

condition is satisfied, so bearing is operated. on hydre dynamic lubrication.

step 6: Coefficient of friction

From McKEES Equation

$$\mu = \frac{33.25}{10^{16}} \left(\frac{20}{P}\right) \left(\frac{D}{C}\right) + \kappa$$

C-diameteral cleanance (Refer DB pg. 7.32)

C = 125 to 200 micron for D=150 mm

Take 
$$C = 150 \text{ micron}$$
 $= 150 \times 10^{-5} \text{ m}$ 
 $= 15 \times 10^{-3} \text{ cm}$ 
 $K = 0.002$ 
 $E = 33.25 = (2844.5) = (15 \times 10^{-3}) = (10^{-5})$ 

Blop 9: Antifical Cooling is required or not?

Ity = 20-18 ×10<sup>3</sup> kgfm/min

Hd = 0.084 kgfm/min

Hg > Hd, Antifical cooling is required.

Step 10: Mass of oil (m)

m=Hg-Ha C'Oto

Assume  $\Delta t_0 = 10^{\circ} C$   $m = 20.18 \times 10^3 - 0.084$   $171 \times 10$ 

m =11.80 kgf/min.

Step 11: Belection of Bearing material (Refer DB Pg . 7.30)

Rubber material is selected for the the application of marine propellors, pumps,

Turbine:

Step 12: Diameter of Bearing.

 $D_b = D + C$ =  $15 + 15 \times 10^{-3}$ 

=15.015 cm.

2. Design a Journal bearing for certifugal pump from the following data

food on the Journal = 20,000 N

Speed of the Journal = 900 pm

Type of oil SAE = 10 for which absolute viscosity at 55°C = 0.017 kg/m sec, ambient temperature of oil =15°C, Maximum bearing pressure post the pump = 1.5 N/mm2. calculate also mass of the lubricating oil nequired for artifical cooling is rise of temperature of oil limited to 10°C. Heat dissipation coefficient 1232 W/m²/°C

Gen Data: -

W = 20,000N = 2000 kgt

N=900 spm

Type of oil SAE =10

to = 55°C

Z = 0.017 kg/msec

=ITCP

ta = 15°C

[P] = 1.5 N/mm<sup>2</sup> = 15 kgf/cm<sup>2</sup>

DE =100

C = 1232 W/m2/0c

To Find! -

mass of oil (m)

Bolution! -

Stop! Dimension of Journal bearing (7.31) For centrifugal pump

$$\frac{L}{D} = 1 - 2$$

Take 1 = 1.5

Diameter is not given assume D = 100 mm D = 100 cm

1=1.5 XID

L = 15 cm

Ostep 2 pressure developed (or) check for bearing pressure.

[P] = 15 kgf/cm² (Given)

PJind - W DXL

= <u>Jooo</u>

= 13.33 kg f/cm2

PJind C[PJall, Design is safe.

Step 3: Viscosity of oil

Z = 0.017 kg/ms (Given).

=17 CPS

Step 4: Selection of grade

Oil SAE = 10 (Given)

Step 5: Bearing module.

From Basic theory Zn] > 1 [Zn]
P Jud

 $\frac{17 \times 900}{13.33} > \frac{1}{3} (2844.5)$ 

1147.7 > 948.16

condition is is satisfied so the bearing is operated on hydro dynamic Lubrication.

Step 6: Coefficient of friction <7.34> From MckEES Equation  $\mu = \frac{33.25}{10^{10}} \left( \frac{20}{P} \right) \left( \frac{D}{C} \right) + \kappa$ c-diameteral clearance <7.32> C = 100 to 175 microns for D = 100 mm C = 150 micron = 150 X10 m c = 15 x10 cm. K = constant 20.0025  $H = \frac{33.25}{10^{10}} (2844.5) \left( \frac{16}{15 \times 10^{-3}} \right) + 0.0025$ M = 8.79 x10-3 Step 7: Heat generated (or) Power loss Ha = HWV kgf m/min = 8.79×10 ×2000 × TTXNO ×900. =4.97 X103 kgfm/min. Step 8: Heat dissipated Uy = (A+ +18)2LD Hy = CA [to-ta] = CXLXD [to-ta] = (20-18) × 0.15 × 0.

= 1232 XO.15 XO.10 (20) H = 369-6 W

-0.049 NOT MA

Step 9: Artifical coolings is required or not.  $Hg = 4.97 \times 10^{3} \text{ kgf m/min}$   $= 4.97 \times 10^{2} \text{ W} - 228.32 \text{ W}$  Hd = 369.6 W

Hg > Hf, Artifical cooling is required.

Step 10: mass of ail (m)

 $m = \frac{+b_1 - +b_2}{c'ot_0}$ 

c' = 171 kgf m/kgf°c

 $m = \frac{4.97 \times 10^3 - 36960.049}{171 \times 10}$ 

S

m = 0.26 kgd/min

Stubble 1

Design of Journal bearing Subjected to Radial load Hydro dynamic bearing (360°, 180°, 120°, 60°).

DESIGN PROCEDURE ! -

Step 1: Dimension of Journal bearing

 $\frac{L}{D} = ?$   $\angle 7.317$ 

\* If machine is not given assume centrifugal pump.

\* Diameter of Journal is not given

Assume D = 100mm.

\* Power is given  $p = 2\pi NT$ 60 × 1000

\* Torque  $T = \frac{\pi}{16} T D^3$ ; D = ?

Step 2: Prussure developed (Or) Check for bearing prussure.

in kgt/cm²

induced bearing possessure,  $Pind = \frac{Load}{A} = \frac{W}{LxD}$ 

W-load in kgt

LID in cm

Pind <[P]all

Design is safe.

step 3: Viscosity of oil.  $Z = ? \qquad \angle 7.31 >$ 

Step 4! Sommer field number

47.347

 $S = \left(\frac{Z'n'}{P}\right) \left(\frac{D}{C}\right)^2$ 

where, P = Pirducad

z'- absolute viscosity

 $z' = \frac{z}{9.81 \times 10^7}$  kgf Sec/cm<sup>2</sup>.

n'-speed in sups.

D\_Diameter of Journal

C-Diameteral cleavance

C = 2 × radial clearance

= 2 × Cx

step 5: Dimensionless performance parameter

(Ruber DB Pg. 7.36, 7.37,

\* Coefficient of friction variable. 7.38, 7.39)

= H B

\* Flow variable = 49 DCn'L

\* Flow realtie = 95

\* Pressure quatio - P.
Proax

where, P = Pinducad

\* Temperature rise variable = PC 1to

\* Minimum oil film thickness = 2ho

step 6: Heat generated (or) power loss in

Hg = HWV in kgf m/min

Stop 7: Heat dissipated Hy

 $Hd = \frac{(\Delta t + 18)^2 LD}{k}$  kgf m/min

Ha = CA Ato.

 $\Delta t_0 = t_0 - t_a$ 

Step 8: Amount of contifical cooling required

= Hg - Ha

Stop 9: mass of lubricating oil.

 $m = \frac{Hg - Hd}{c' \Delta t_0}$  kgf/min.

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